

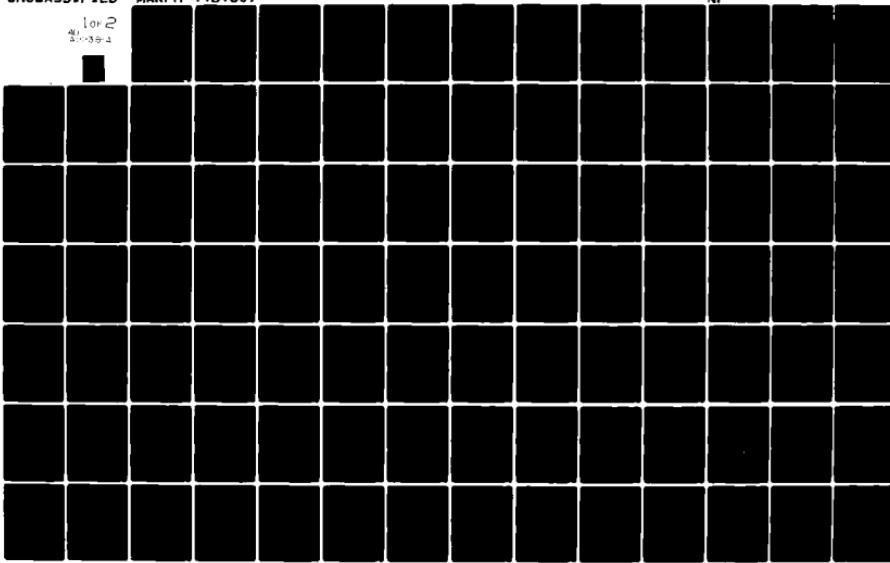
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UNITED STATES AIR FORCE

HILL AIR FORCE BASE, UTAH 84056

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**PROPELLANT**  
**SURVEILLANCE REPORT**  
**LGM-30 A&B STAGE I,**  
**TP-H1011.**

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PROPELLANT ANALYSIS LAB

MAKPH REPORT

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NOVEMBER 1980

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PROPELLANT SURVEILLANCE REPORT

LGM-30 A & B STAGE I

TP-H1011

Author

*John A. Thompson*  
JOHN A. THOMPSON, Chemist  
Component & Combustion Test Unit

Engineering & Statistical Review By

*John K. Scambia*  
JOHN K. SCAMBIA, Project Engineer  
Service Engineering

*Edward J. Erickson*  
EDWARD J. ERICKSON, Statistician  
Data Analysis Unit

*Ronald F. Larsen*  
RONALD F. LARSEN, Chief  
Physical & Mechanical Test Unit

*Leonidas A. Brown*  
LEONIDAS A. BROWN, Chief  
Component & Combustion Test Unit

Approved By

*Anthony J. Inverso*  
ANTHONY J. INVERSO, Chief  
Propellant Analysis Laboratory



November 1980

Missiles, Airmunitions, and Weapons Division  
Directorate of Maintenance  
Ogden Air Logistics Center  
United States Air Force  
Hill Air Force Base, Utah 84056

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## ABSTRACT

This report contains propellant test results from cartons of TP-1011 bulk propellant representing LGM-30 A and B First Stage Minuteman Motors. Testing was accomplished in accordance with MMWRBM Project MD4046C-WNL01529.

The purpose of testing was to determine and provide early warning of any serious degradation trends occurring in the propellant for service life predictions.

An analysis of all parameters indicates that no potential problems are expected in the propellant for at least two years past the oldest data point.

Data stored in the G085 System were plotted utilizing the IBM 360-65 Computer and CAL-COMP Plotter. The data range at any age can be found by suitable inquiry of the G085 System.

Each point on the regression plot represents the mean of all samples at that particular age. The number of specimens at each point is indicated on the sample size summary sheet accompanying each regression plot or group of regression plots.

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DATA TAB	1	1
Unclassified	1	1
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Availability Codes		
Dist	Area only or	
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29B	Zero Time Test Results	29 Jan 64
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300	Minuteman Stage I Motor Reliability Improvement Program Surveillance	May 74

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302	Propellant Surveillance Report LGM-30	Nov 74
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337	Propellant Surveillance Report LGM-30 A & B, Stage 1, TP-H1043	Feb 76
339	Stage 1, New MAPO & ERL-510 Qualification	Mar 76
341	Propellant Surveillance Report LGM-30 Dissected Motors, Phase VII, TP-H1011	Mar 76

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345	Propellant Surveillance Report LGM-30 F & G, Stage 1 Phase B, Series III, TP-H1011	Jun 76
350	Qualification of a New MAPO Source and ERL-510 Curing Agent for Minuteman, Stage 1, UF-2121 Liner	Sep 76
351	Propellant Surveillance Report LGM-30 A & B, Stage 1, TP-H1011	Sep 76
354	Minuteman Stage 1 Motor Reliability Improvement Program Surveillance	Sep 76
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367	Propellant Surveillance Report LGM-30 A & B, Stage 1, TP-H1011	Apr 77
370	Propellant Surveillance Report LGM-30 F & G, Stage 1, Phase E, Series II, TP-H1011	Apr 77
377	Qualification of a New MAPO Source and ERL-510 Curing Agent for Minuteman Stage 1, UF-2121 Liner	Oct 77
379	Final RIP Report, Minuteman Stage 1 Motor Reliability Improvement Program Surveillance	Oct 77
385	Propellant Surveillance Report LGM-30 A, B, F, & G, Stage 1, TP-H1043	Dec 77
388	Propellant Surveillance Report LGM-30 A & B Stage 1, TP-H1011	Jan 78
390	Propellant Surveillance Report LGM-30 F & G Stage 1, Phase E, Series IV, TP-H1011	Feb 78
392	Propellant Surveillance Report LGM-30 Dissected Motors, Phase IX, TP-H1011	Mar 78
393	Propellant Surveillance Report LGM-30 A & B Stage 1, TP-H1011	May 78

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<u>Report Nr</u>	<u>Title</u>	<u>Report Date</u>
396	Propellant Surveillance Report LGM-30 F & G Stage I, TP-H1011	Jun 78
405	Propellant Surveillance Report LGM-30 F & G Stage I, TP-H1011	Oct 78
406	Propellant Surveillance Report LGM-30 Dissected Motors, Phase X, TP-H1011	Nov 78
416	Propellant Surveillance Report LGM-30 F and G Stage I, TP-H1011	Apr 79
423	Propellant Surveillance Report LGM-30 F and G Stage I, TP-H1011	Oct 79
424	Propellant Surveillance Report LGM-30 Stage I, TP-H1043	Nov 79
425	Propellant Surveillance Report LGM-30 A and B Stage I, TP-H1011	Nov 79
427	Propellant Surveillance Report LGM-30 Dissected Motors, Phase XI, TP-H1011	Nov 79
438	Propellant Surveillance Report LGM-30 F and G Stage I, TP-H1011	Apr 80
445	Propellant Surveillance Report LGM-30 F and G Stage I, TP-H1011	Sep 80

## GLOSSARY OF TERMS AND ABBREVIATIONS

Aging Trend	A change in properties or performance resulting from aging of material or component
CSA	Cross Sectional Area
DB	Dogbone
Degradation	Gradual deterioration of properties or performance
E	Modulus (psi), defined as stress divided by strain along the initial linear portion of the curve.
EB	End Bonded
EGL	Effective Gage Length
em	Strain at maximum stress
er	Strain at rupture
"F" ratio	The ratio of the variance accounted for by the regression function to the random unexplained variance. The regression function having the most significant "F" ratio is used for plotting data. The ratio is also used in detecting significant changes in random variation between succeeding time points
JANNAF	Joint Army, Navy, NASA, Air Force
MAKPH	Propellant Lab Section at Ogden Air Logistics Center
Ogden ALC	Ogden Air Logistics Center, Air Force Logistics Command
r or R	The Correlation Coefficient is a measure of the degree of closeness of the linear relationship between two variables
Regression Equation	The general form of the regression equation is $Y = a + bx$
Regression Line	Line representing mean test values with respect to time
$s_b$	Standard error of estimate of the regression coefficient

### GLOSSARY OF TERMS AND ABBREVIATIONS (cont)

$S_e$ or $S_{Y.X}$	Standard deviation of the data about the regression line
$S_m$	Maximum Stress
$S_r$	Stress at rupture
Standard Deviation ( $S_y$ )	Square root of variance
Strain Rate	Crosshead speed divided by the EGL
"t" test	A statistical test used to detect significant differences between a measured parameter and an expected value of the parameter (determines if regression slope differs from zero at the 95% confidence level)
Variance	The sum of squares of deviations of the test results from the mean of the series after division by one less than the total number of test results
3 Sigma Band	The area between the upper and lower 3 sigma limit. It can be expected that 99.73% of the inventory represented by the test samples would fall within this range assuming that the population is normally distributed.
90-90 Band	It can be stated with 90% confidence that 90% of the inventory represented by the test samples would fall within this range assuming that the population is normally distributed

## INTRODUCTION

### A. PURPOSE:

Quality assurance tests have been conducted for seventeen years on First Stage LGM-30A and B Minuteman Motor Propellant blocks to evaluate the effects of aging on TP-H1011 propellant.

Statistical analysis of the tests performed, as directed by Engineering, should provide early warning if serious degradation trends occur. Annual evaluation of the propellant provide data that can be directly input into engineering reliability and service life predictions. Testing was performed in accordance with MMWRM Directive GTD-1C and GTD-1C Amendments 1 and 2.

### B. BACKGROUND:

Testing was first accomplished at MANCP on LGM-30A TP-H1011 propellant blocks in 1963 and was designated Zero-Time Testing (MAGCP Report Nrs 29B, 29C and 29F). Subsequent testing was accomplished at approximately 24 month intervals (MAGCP Report Nrs 29G, 29H - Phase I; 76 - Phase II; 181 - Phase III).

LGM-30B Zero-Time testing was accomplished in 1964 with subsequent testing at intervals of 24 months (MAGCP Report Nrs 32A-Zero-Time; 32C, 49, 53, 55, 58, 61, 66 - Phase I: 118, 126, 130-Phase II; 195, 268 - Phase III).

Reports prior to MAGCP Report Nr 223(72) contained raw data using sigma relation to compare to Zero-Time variance. MANCP Report Nr 239(72) published in April of 1972 contained all the

data on LGM-30 A, B, F and G in the G085 System at that time. Report Nos. 258(72) and 268(73) reported LGM-30 A and B data in statistical analysis by itself. This report is the eleventh time that LGM-30 A and B data have been reported in this manner.

Zero time testing was started as soon as possible after receipt of the propellant by MAKPH. Data from these tests were used to establish a base line for each test to which each subsequent test data (ATP - Accelerated Test Program) were compared in the reports listed above.

The LGM-30 A and B test matrix (Table 1) was used to determine the number of specimens to be taken from each propellant loaf and the specific test or tests to which these specimens were subjected. Low rate tensile and hardness specimens were taken from all LGM-30 A and B blocks. Specimens for other physical and combustion tests were taken from every seventh block.

The testing program has been revised over the years. Some tests have been added later in the program. Therefore, early test data in the 0 - 6 year period may not always be available for inclusion in a particular regression analysis.

Table 1  
Test Program

The test matrix is taken from GTD-1C, Amendment 2, and the tests, conditions, number of specimens and test methods are listed below.

<u>Test</u>	<u>Conditions</u>	<u>Description</u>	<u>Per Cond</u>
Hardness	10 Sec	Dogbone Ends	3
Low Rate Tensile	2.0 in/min	1/2" JANNAF Dogbone	3
High Rate Tensile	1750 in/min	3/4" Dogbone	3
High Rate Triaxial Tensile	600 psi, 1750 in/min	3/4" GL Rail End Bonded	1
Low Rate Biaxial Tensile	0.2 in/min	3/4" GL Rail End Bond	1
Stress Relaxation	3% & 5%	1/2" x 1/2" x 4" EB	3
Dynamic Response	70 gm ct wt	3.3" dia x .33" disc	1
Sol Gel		1/2" x 1/2"	8
VLR	$2 \times 10^{-3}$ in/min	1/2" JANNAF Dogbone	3
Ignitability	168 cal/cm <sup>2</sup> sec	.050" wafer	3
TCLE		.200" wafer	3
Pressure Time	500 psi	1/2" x 3/8" x 1"	3
Burning Rate	1000 psi	.156" x .156" x 5" Strand	3
DTA	12°C Rise/min	.040" wafer	3
DSC		.040" wafer	3
Poisson's Ratio	77°F $\pm$ 2° 15% Strain	.50" x .50" x 4"	6
Tear Energy	70°F $\pm$ 2°	0.1" x 1.18" x 3"	6
Failure Envelope		JANNAF Dogbone	3

## STATISTICAL APPROACH

In order to determine aging trends for shelf/service life predictions, as directed by Service Engineering, First Stage LGM-30 A and B Minuteman Motor propellant (TP-H1011) blocks have been under-going testing since 1963, statistically analyzed and reported on a regular test cycle by this laboratory.

The primary reason for performing statistical analysis on test data is for the detection of propellant changes due to aging that would affect motor reliability. Regression analysis was the method used to examine data and to aid in drawing conclusions about dependency relationships that may exist i.e., relationship between age versus test results.

The linear model ( $Y = a + bX$ ) was found to be the best fit model for almost all regressions within this report. The only exception was for the test parameter pressure time at maximum pressure. For this parameter the reciprocal of  $X$  ( $Y = a + b(1/X)$ ) proved to be the best fit model.

Individual data points from different time periods were used to establish a least squares trend line for the data. The variance about the regression line, obtained using individual values of the dependent variable, was used to compute a tolerance interval such that at the 90% confidence level 90% of the sample distribution falls within this interval. This tolerance interval was extrapolated to a maximum of 24 months into the future from the age of the oldest motor tested. The 't' values and the significance of this statistic, which are reported for each regression model, give an indication of the 'statistical significance' of the slope of the trend line as compared to a line of zero slope.

Data were plotted by computer. The 'y' axis is computed so that the values at one inch intervals are peculiar to the data spread of the parameter tested. Plotted data points represent means at the particular ages at which testing occurred. The number of specimens at each age point is indicated on the sample size summary sheet accompanying each regression plot or group of regression plots. Variance at each test age can be determined by consulting the G085 data storage system.

In a few cases, a small change has become apparent in data variance and regression trend lines. However, the changes are gradual and no operational problems are expected at this time.

A small post-cure effect (propellant properties are affected by curing reactions during the first year or two after manufacture) has been observed on some of the earliest test data. This curing effect tends to bias and skew the projected trend lines. To reduce this bias, some of the earliest data has been deleted from the regressions. As the data from the total sample population increases in size, this deleted data causes no problems. By compensating for this post-cure biasing, a more accurate aging trend line for service life prediction is provided.

## TEST RESULTS

### A. TENSILE:

Regressions for very low rate tensile data show a statistically significant decrease in the strain parameters. The stresses and modulus regressions show a statistically significant increase (Figures 1 thru 5).

Low rate strain regressions show a statistically significant decrease. The stresses and modulus regressions show a statistically significant increase (Figures 6 thru 10).

For low rate biaxial tensile testing, the strain at rupture does not show a significant trend. The strain at maximum stress, stresses, and modulus regressions show a statistically significant increase (Figures 11 thru 15).

The high rate tensile strain at maximum stress shows a statistically significant increase. Maximum stress has a non-significant aging trend. Strain at rupture shows a statistically significant decrease. Stress at rupture and modulus show a statistically significant increase (Figures 16 thru 20). The large variation in data points for modulus at 12 to 16 1/2 years is mostly due to a reduced number of specimens tested per month.

For triaxial tensile testing, the strains and stresses show a statistically significant increase. Modulus shows a statistically significant decrease (Figures 21 thru 25).

For all of the tensile testing, where changes are shown, the trends are gradual and no operational problems are expected for at least two years beyond the oldest data point.

**B. STRESS RELAXATION:**

Modulus at both 3% and 5% strains show a statistically significant increase for all time periods (Figures 26 thru 33). However, the slopes of the trend lines are gradual and no operational problems with the propellant are expected.

**C. DYNAMIC RESPONSE:**

The storage shear modulus at 200 and 400 Hz shows a statistically significant decrease while the loss tangent at 200 and 400 Hz shows a statistically significant increase (Figures 34 thru 37).

**D. CONSTANT STRAIN:**

Strain at rupture for constant strain does not show a significant change (Figure 38).

**E. TCLE (THERMAL COEFFICIENT OF LINEAR EXPANSION):**

The thermal coefficient of linear expansion below and above the glass transition point shows a statistically significant increase (Figures 39 and 40).

**F. SOL GEL:**

The cross link density shows a statistically significant increase with a statistically significant decrease for percent extractables and

for weight swell ratio a non-significant change (Figures 41 thru 43).

The increasing cross link density trend correlates well with the other physical properties. The tensile testing shows an increase in maximum stress and modulus with strains decreasing. The stress relaxation, dynamic response and constant strain results also correlates well with cross link density.

G. DTA (DIFFERENTIAL THERMAL ANALYSIS):

For the DTA regressions the endotherm and first and second exotherms show a statistically significant decrease. The third exotherm shows a statistically significant increase. There is no significant change for ignition temperature (Figures 44 thru 48).

H. PRESSURE TIME:

Maximum pressure shows a statistically significant decrease and the time to maximum pressure shows a statistically significant increase (Figures 49 and 50):

I. BURNING RATE:

The burning rate shows a statistically significant decrease (Figure 51). This correlates with the increasing time to maximum pressure.

J. IGNITABILITY:

Ignitability shows no significant change (Figure 52).

K. DSC (DIFFERENTIAL SCANNING CALORIMETER):

The endotherm, first and second exotherm show no significant changes (Figures 53 thru 55).

L. TGA (THERMALGRAVIMETRIC ANALYSIS):

The percent weight loss at ignition, ignition temperature, and percent weight loss at 250°C hold, all show a statistically significant increase (Figures 56 thru 58).

In general, where statistically significant changes are shown for the thermal analysis the trends are gradual. From this it is concluded that the propellant will remain thermally stable for at least two years beyond the last data point.

## CONCLUSIONS

This report includes LGM-30 A and B bulk propellant test results presently in the G085 System and covers the past seventeen years of testing.

The test results show that under present storage conditions the physical/mechanical and combustion properties of the propellant are remaining relatively stable with age. This is indicated by the regression plots where the slope of the trend line is relatively flat or close to a line of zero slope and have not changed appreciably from the last test period.

From the statistical analyses, all tests conducted indicate that motor propellant reliability will not be affected for at least two years past the last data point on the regression.

AGE (MOS)	NR SAMP								
15	18	96	29	121	24	147	3	203	3
17	21	97	34	122	21	152	6	204	6
18	6	98	38	123	19	153	24	205	9
21	6	99	33	124	20	154	12	206	14
73	5	100	34	125	18	155	21		
75	18	101	24	126	51	156	5		
77	18	102	22	127	26	157	4		
78	3	103	40	128	42	158	3		
79	9	104	20	129	28	160	6		
80	30	105	39	130	18	162	6		
81	18	106	28	131	25	163	6		
82	9	107	28	132	24	164	3		
83	18	108	44	133	20	165	3		
84	18	109	22	134	39	166	3		
85	6	110	26	135	45	168	6		
86	40	111	24	136	18	169	3		
87	27	112	29	137	39	185	1		
88	23	113	41	138	21	191	11		
89	13	114	15	139	12	192	10		
90	7	115	16	140	24	193	9		
91	15	116	27	141	33	194	3		
92	7	117	12	142	21	195	6		
93	12	118	33	143	6	196	3		
94	19	119	27	145	3	197	6		
95	22	120	36	146	3	199	3		

STAGE 1, WING 162 VERY LOW RATE CRSS=0.002 IN/MIN STRAIN AT MAX STRESS (EM)

This sample size summary is applicable to figures 1 Thru 5

$F = +1.3997039E+01$     $R = -8.5315665E-02$     $L = +3.7412617E+00$     $N = 1911$   
 $S_r =$  SIGNIFICANT    $S_a =$  SIGNIFICANT    $S_l =$  SIGNIFICANT  
 $S_f =$  SIGNIFICANT    $S_b =$  SIGNIFICANT    $S_t =$  SIGNIFICANT  
 $Degrees of Freedom = 1909$   
 STORAGE CONDITIONS = AMB TEMP/RH   TEST CONDITIONS = 77 DEG/F AMB-RH

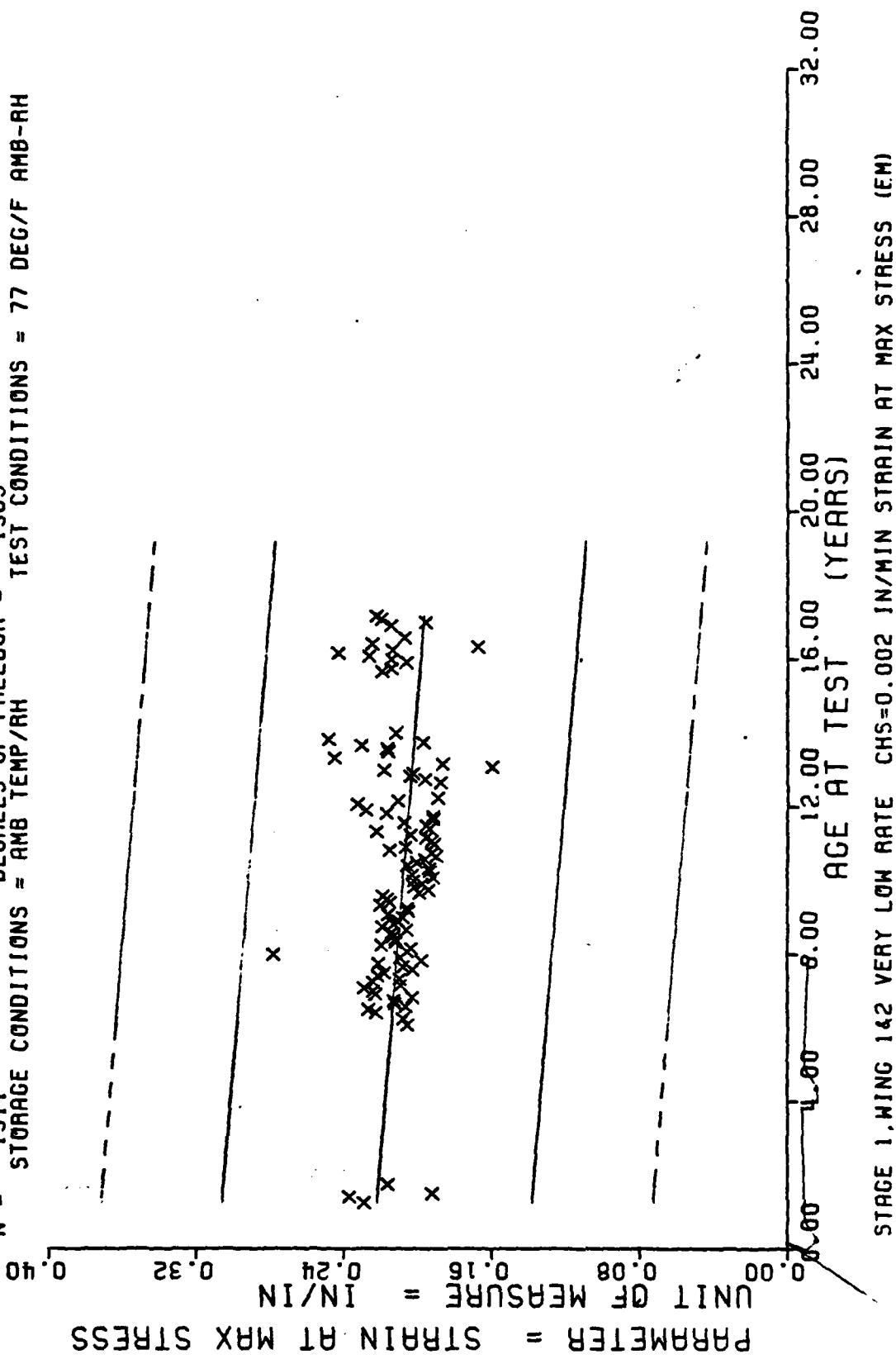


Figure 1

$\gamma = (1 + 8.0885938E+01) + (1.7539642E-02) \times X$   
 $F = +8.2357028E+00$  SIGNIFICANT  
 $R = +6.5540927E-02$  SIGNIFICANT  
 $I = +2.8697914E+00$  SIGNIFICANT  
 $N = 1911$  DEGREES OF FREEDOM = 1909  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

UNIT OF MEASURE = PSI  
 PARAMETER = MAXIMUM STRESS

STAGE 1. WING 142 VERY LOW RATE CHS=0.0002 IN/MIN MAXIMUM STRESS (SM)

Figure 2

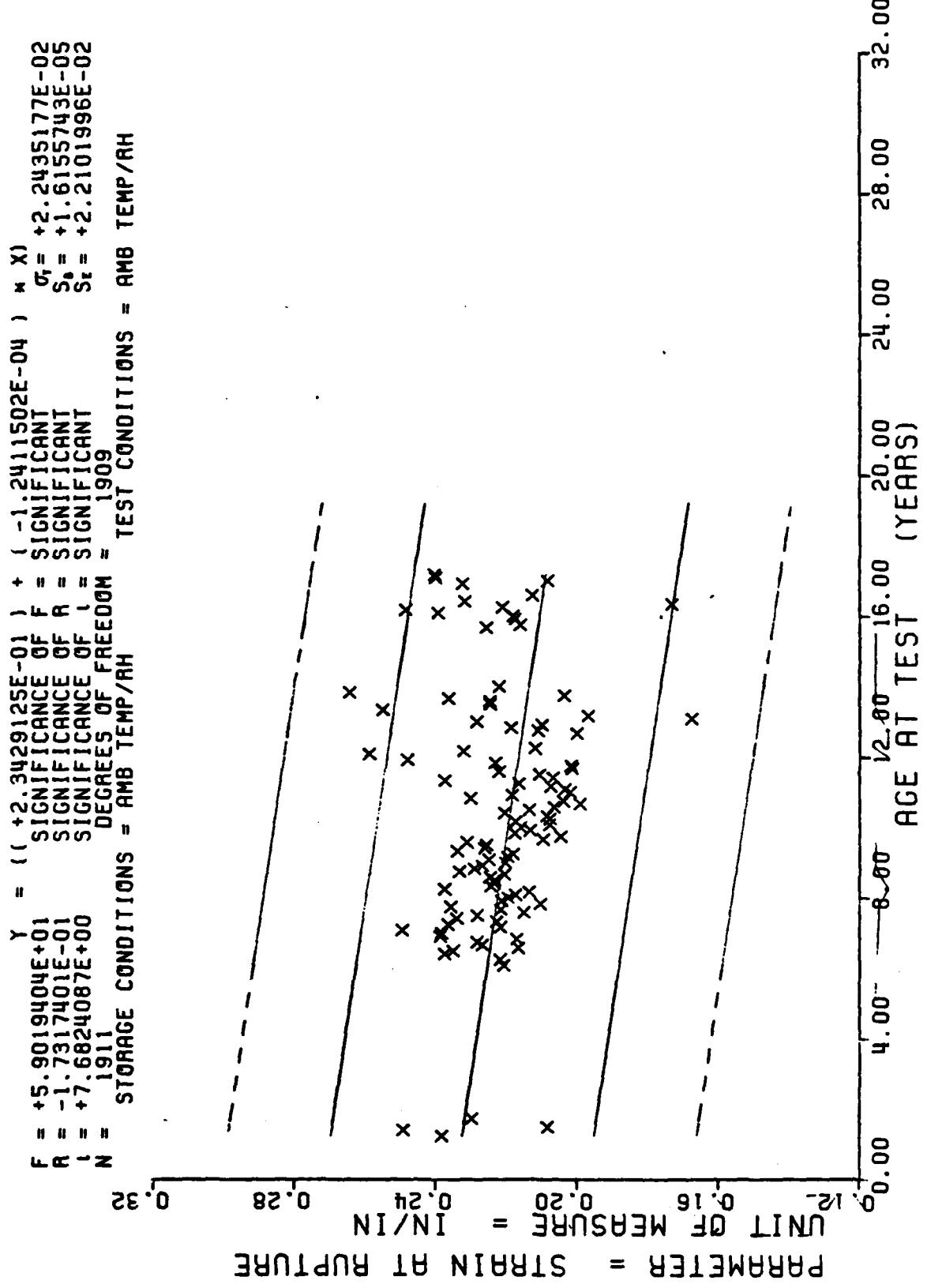


Figure 3

$Y = (( +7.5536754E+01) + (+2.1059042E-02) * X)$   
 $F = +8.8251166E+00$  SIGNIFICANCE OF F = SIGNIFICANT  
 $R = +6.7835296E-02$  SIGNIFICANCE OF R = SIGNIFICANT  
 $I = +2.9707097E+00$  SIGNIFICANCE OF I = SIGNIFICANT  
 $N = 1911$  DEGREES OF FREEDOM = 1909  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

PARAMETER = STRESS AT RUPTURE  
 UNIT OF MEASURE = PSI

STAGE 1. WING 142 VERY LOW RATE CHS=0.0002 IN/MIN STRESS AT RUPTURE (SR)

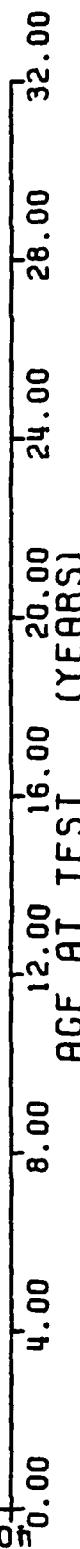
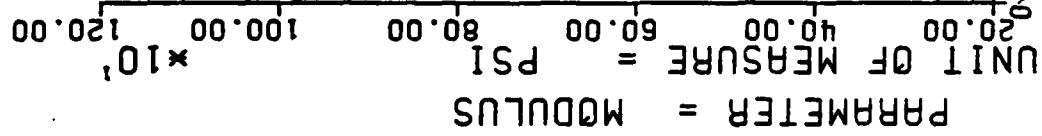


Figure 4

$F = +7.2393650E+00$  SIGNIFICANCE OF  $F =$  SIGNIFICANT  
 $R = +6.1480686E-02$  SIGNIFICANCE OF  $R =$  SIGNIFICANT  
 $I = +2.6906068E+00$  SIGNIFICANCE OF  $I =$  SIGNIFICANT  
 $N = 1910$  DEGREES OF FREEDOM = 1908

STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = 77 DEG/F AMB-RH



STAGE 1. WING 1&2 VERY LOW RATE CHS=0.002 IN/MIN MODULUS (E)

Figure 5

*** SAMPLE SIZE SUMMARY ***									
AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
1	1	30	252	55	24	81	156	106	274
6	5	31	307	56	18	82	260	107	153
7	33	32	215	57	30	82	166	108	257
8	52	33	114	58	24	84	162	109	182
9	45	34	91	59	24	85	308	110	165
10	89	35	57	60	42	86	144	111	171
11	40	36	63	62	24	87	472	112	322
12	46	37	56	63	51	88	683	113	155
13	44	38	75	64	81	85	783	114	213
14	65	39	68	65	42	90	505	115	193
15	43	40	54	66	96	91	558	116	204
16	69	41	84	67	79	92	528	117	220
17	42	42	215	68	69	93	297	118	229
18	75	43	242	69	100	94	365	119	162
19	127	44	158	70	89	95	310	120	188
20	65	45	194	71	165	96	212	121	238
21	85	46	136	72	142	97	144	122	74
22	51	47	53	73	59	96	235	123	174
23	55	48	36	74	133	95	215	124	160
24	341	49	27	75	252	100	210	125	198
25	209	50	17	76	153	101	186	126	170
26	251	51	16	77	157	102	174	127	183
27	391	52	21	78	153	102	177	128	156
28	353	53	6	79	134	104	151	129	186
29	617	54	3	80	194	105	217	130	160
									156
									59

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (WOS)	NR SAMPLE	AGE (WOS)	NR SAMPLE
157	65	198	5
158	9	199	9
159	3	202	21
160	24	203	6
161	15	204	15
162	12	205	6
163	39	206	102
164	27	207	59
165	24	208	6
166	30		
167	21		
168	21		
169	5		
184	6		
186	9		
188	5		
189	18		
190	12		
191	106		
192	71		
193	56		
194	30		
195	28		
196	35		
197	12		

$F = +4.3781961E+02$     $Y = ( ( +2.6008970E-01 ) + ( -7.1299136E-05 ) \times X )$   
 $F = \text{SIGNIFICANT}$   
 $R = -1.3270579E-01$     $F = \text{SIGNIFICANT}$   
 $A = +2.0924139E+01$     $R = \text{SIGNIFICANT}$   
 $L = 24425$     $L = \text{SIGNIFICANT}$   
 $N = \text{DEGREES OF FREEDOM} = 24423$     $\text{TEST CONDITIONS} = \text{AMB TEMP/RH}$   
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$

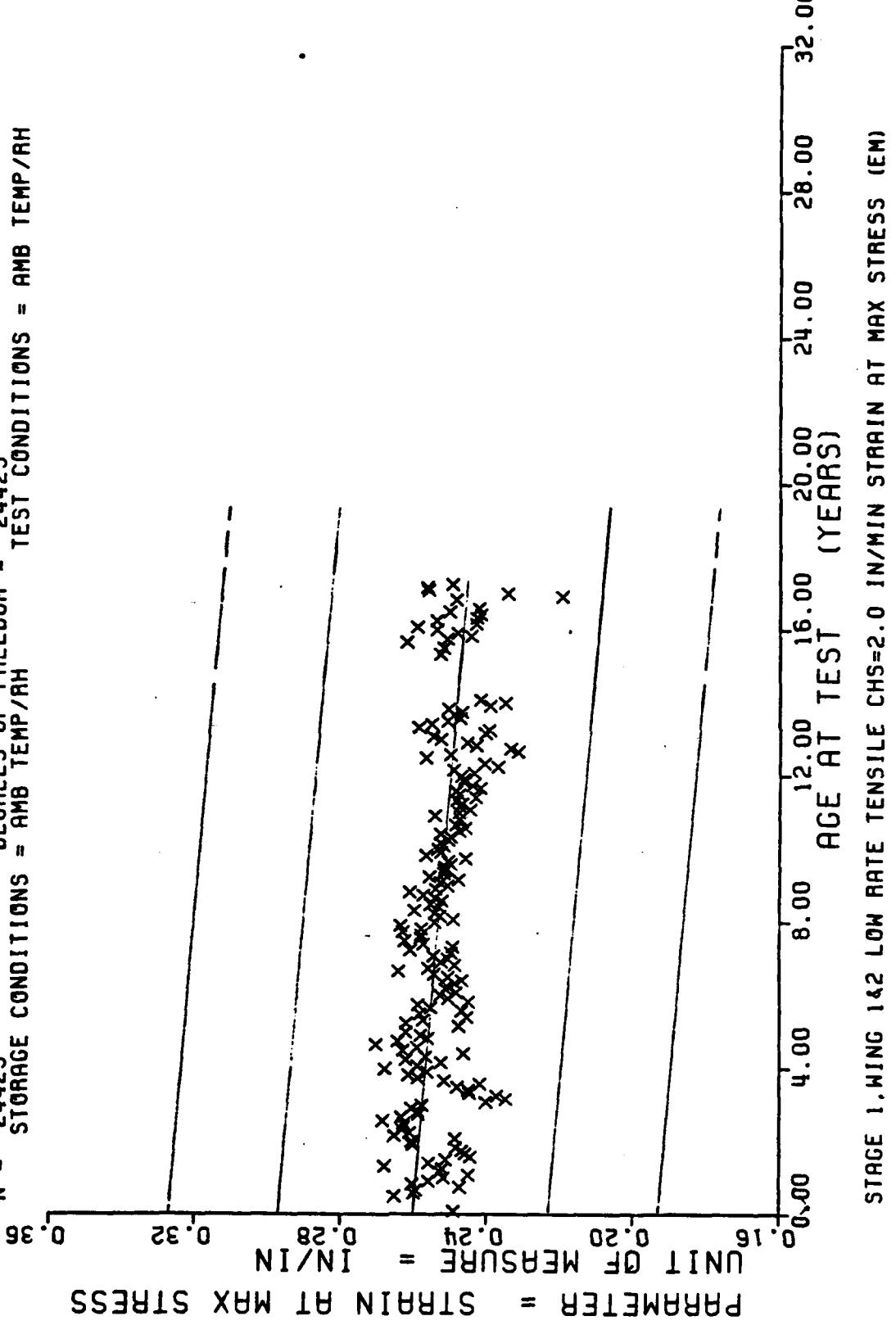


Figure 6

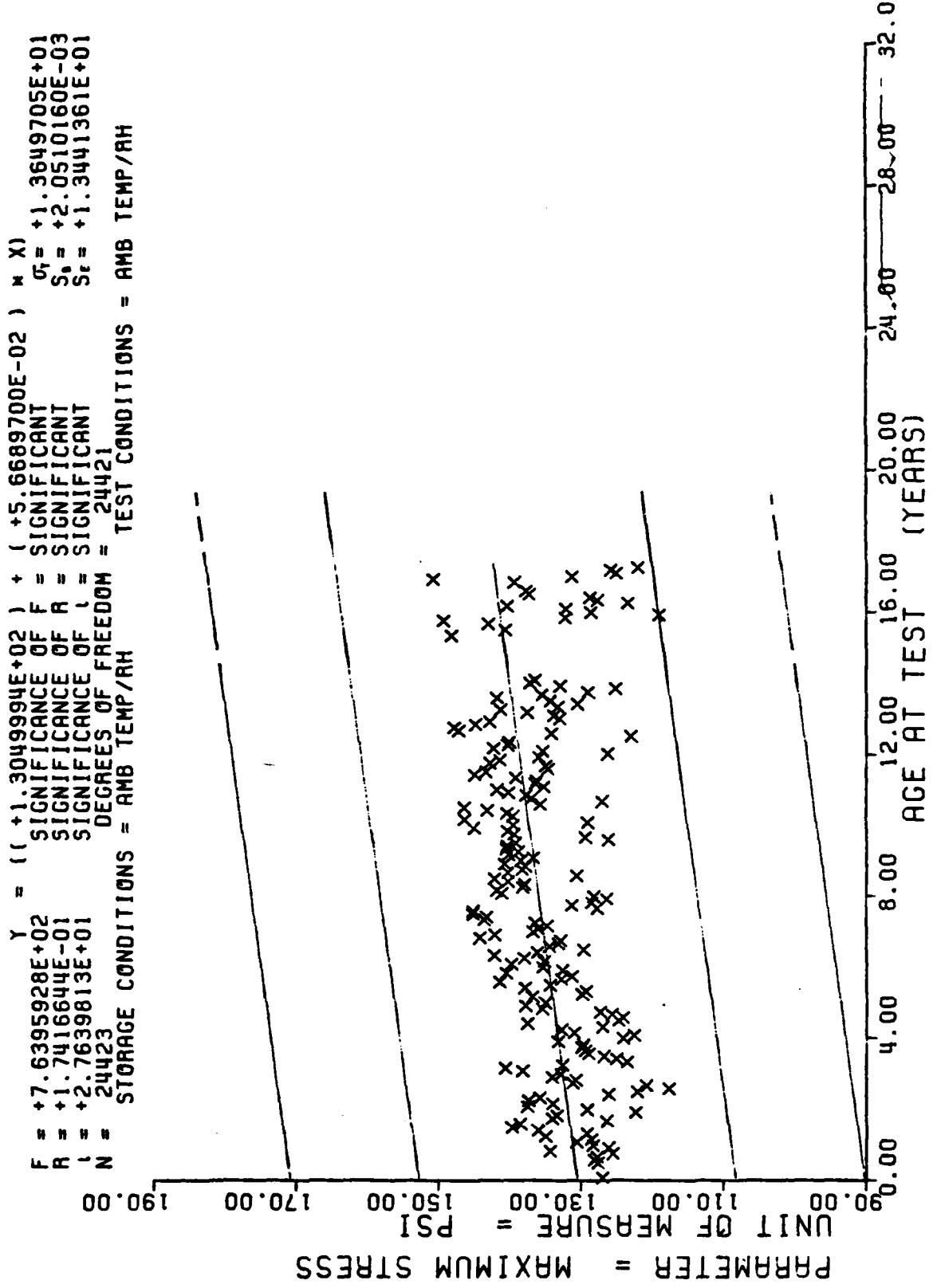


Figure 7

$F = +1.7170034E+03$        $\gamma = (( +3.2369845E-01 ) + ( -1.7012378E-04 ) * X)$   
 $R = -2.5629064E-01$        $F = \text{SIGNIFICANT}$   
 $A = +4.1436739E+01$        $R = \text{SIGNIFICANT}$   
 $I = 24425$        $A = \text{SIGNIFICANT}$   
 $N = 24423$        $I = \text{SIGNIFICANT}$   
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$        $\text{DEGREES OF FREEDOM} = 24423$   
 $\text{TEST CONDITIONS} = \text{AMB TEMP/RH}$

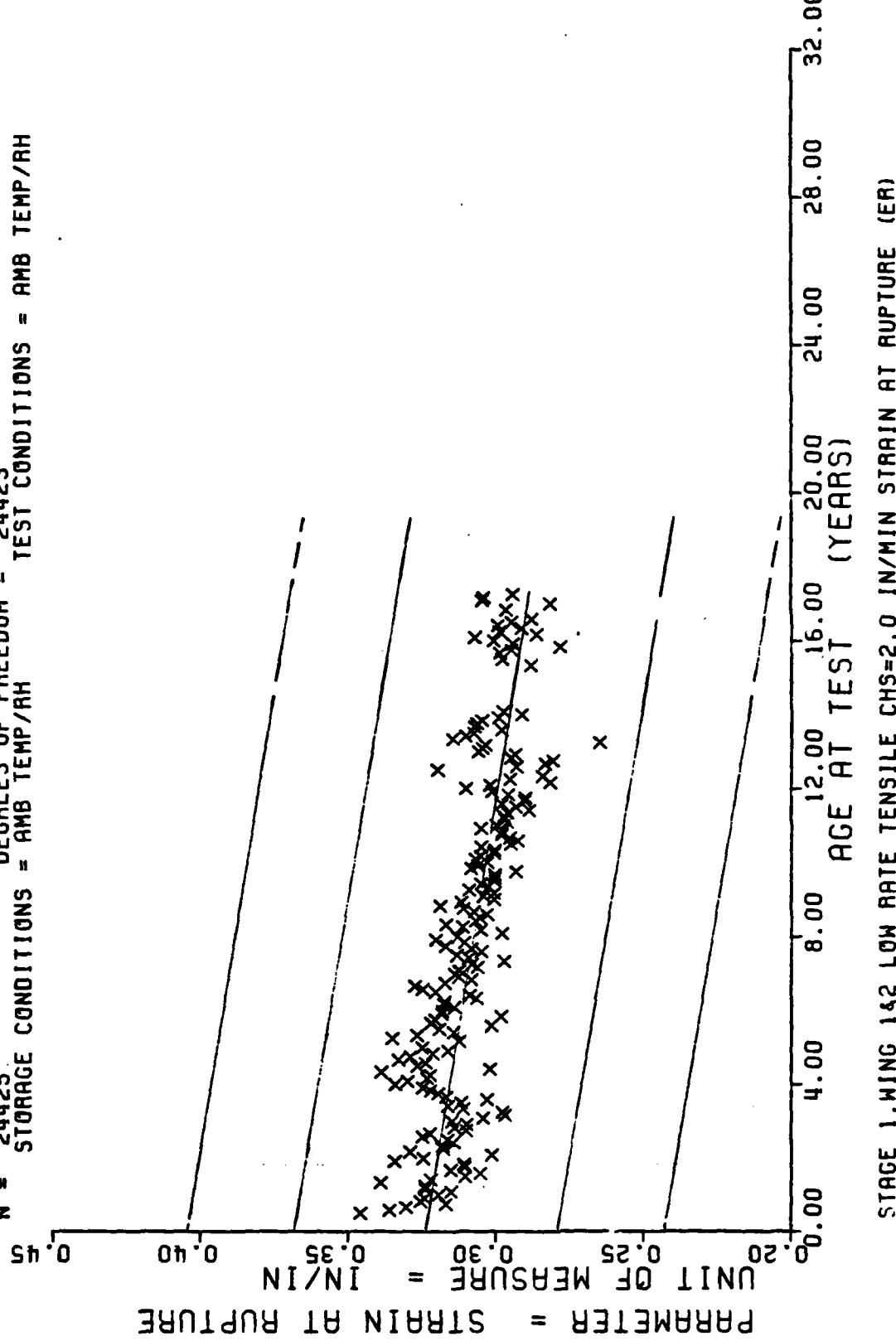


Figure 8

$\gamma = (1 + 1.2346123E+02) + ( + 2.9949969E-02) * X$   
 $F = 1.0201354E+02$  SIGNIFICANCE OF F = SIGNIFICANT  
 $R = +9.0571838E-02$  SIGNIFICANCE OF R = SIGNIFICANT  
 $I = +1.4213146E+01$  SIGNIFICANCE OF I = SIGNIFICANT  
 $N = 24426$  DEGREES OF FREEDOM = 24424  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = 77 DEG/F AMB-RH

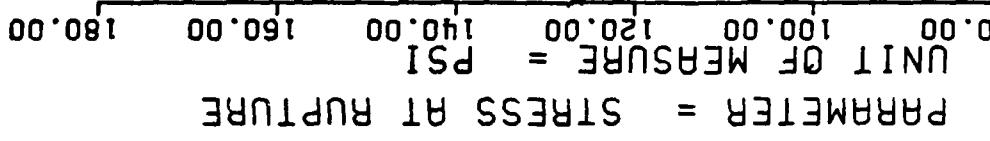


Figure- 9

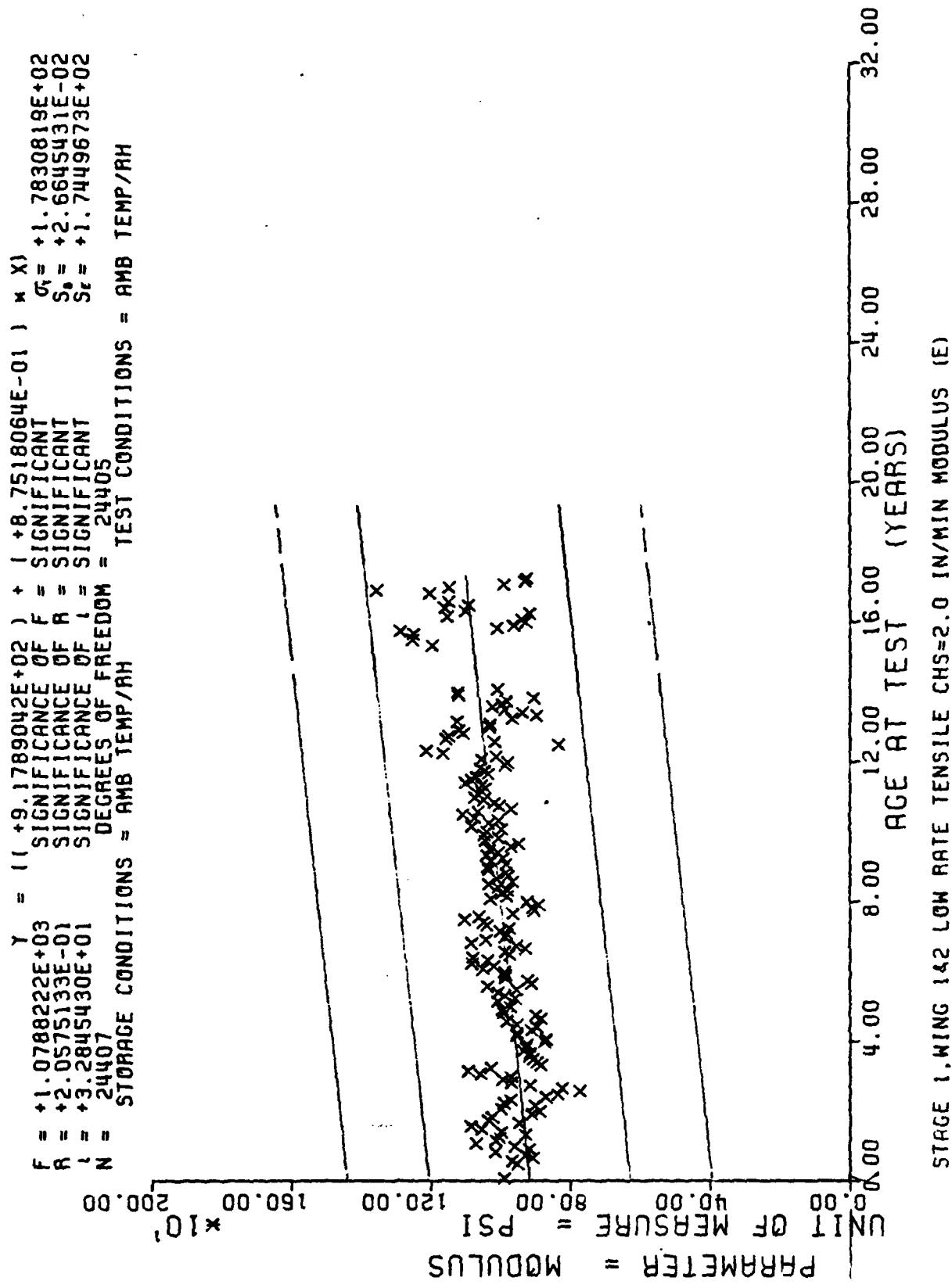


Figure 10

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP										
9	3	31	18	58	2	85	11	110	8	135	13
7	4	32	8	60	1	86	18	111	12	136	16
6	2	33	7	62	2	87	11	112	10	137	10
9	5	34	12	63	1	88	14	113	8	138	8
10	4	35	11	64	1	89	26	114	6	139	1
11	7	36	10	65	1	90	34	115	11	140	2
12	7	37	4	66	2	91	23	116	5	141	9
13	6	38	10	67	2	92	37	117	8	142	5
14	7	39	5	68	1	93	30	118	5	143	1
15	10	40	11	69	2	94	20	119	4	152	1
16	7	41	3	70	2	95	11	120	12	153	5
17	5	43	5	71	4	96	10	121	11	154	3
18	7	44	20	72	3	97	9	122	16	155	5
19	10	45	39	73	9	98	1	123	6	156	6
20	10	46	32	74	4	99	3	124	13	157	12
21	19	47	31	75	5	100	5	125	9	158	6
22	12	48	8	76	9	101	2	126	11	161	2
23	13	49	9	77	12	102	4	127	6	163	1
24	29	50	7	78	7	103	4	128	8	164	3
25	51	51	3	79	10	104	9	129	7	165	1
26	32	52	1	80	5	105	9	130	7	166	3
27	27	53	2	81	10	106	9	131	4	169	1
28	49	54	9	82	15	107	10	132	4	189	2
29	30	55	7	83	13	108	11	133	4	190	2
30	37	56	4	84	10	109	12	134	6	191	8
										192	5
										193	13
										194	1
										195	2
										196	4
										197	3
										201	3
										205	2
										206	2
										207	2

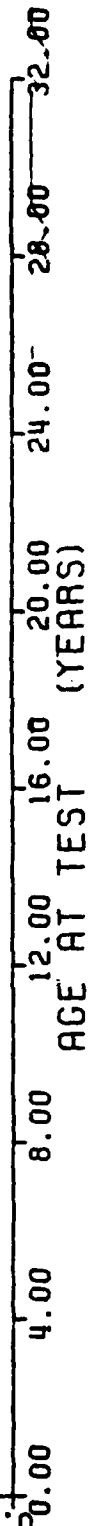
STAGE 1, LOW RATE BIAXIAL CHS=0.02 IN/MIN STRAIN AT MAX STRESS (EM)

This sample size summary is applicable to figures 11 thru 15

$F = +4.3993125E+00$        $Y = (( +2.1073152E-01 ) + ( +2.1762192E-05 ) * X)$   
 $R = +5.4184943E-02$       SIGNIFICANT OF F = SIGNIFICANT  
 $S = +1.0375528E-05$       SIGNIFICANT OF R = SIGNIFICANT  
 $I = +2.0974538E+00$       SIGNIFICANT OF I = SIGNIFICANT  
 $N = 1496$       DEGREES OF FREEDOM = 1494  
 $N =$       STORAGE CONDITIONS = AMB TEMP/RH      TEST CONDITIONS = AMB TEMP/RH

PARAMETER = STRAIN AT MAX STRESS

UNIT OF MEASURE = IN/IN



STAGE 1. WING 1&2 LOW RATE BIAXIAL CHS=0.2 IN/MIN STRAIN AT MAX STRESS (EM)

Figure 11

$F = +5.9490656E+01$    SIGNIFICANCE OF  $F =$  SIGNIFICANT  
 $R = +1.9550196E-01$    SIGNIFICANCE OF  $R =$  SIGNIFICANT  
 $I = +7.7130186E+00$    SIGNIFICANCE OF  $I =$  SIGNIFICANT  
 $N = 1499$    DEGREES OF FREEDOM = 1497

STORAGE CONDITIONS = AMB TEMP/RH

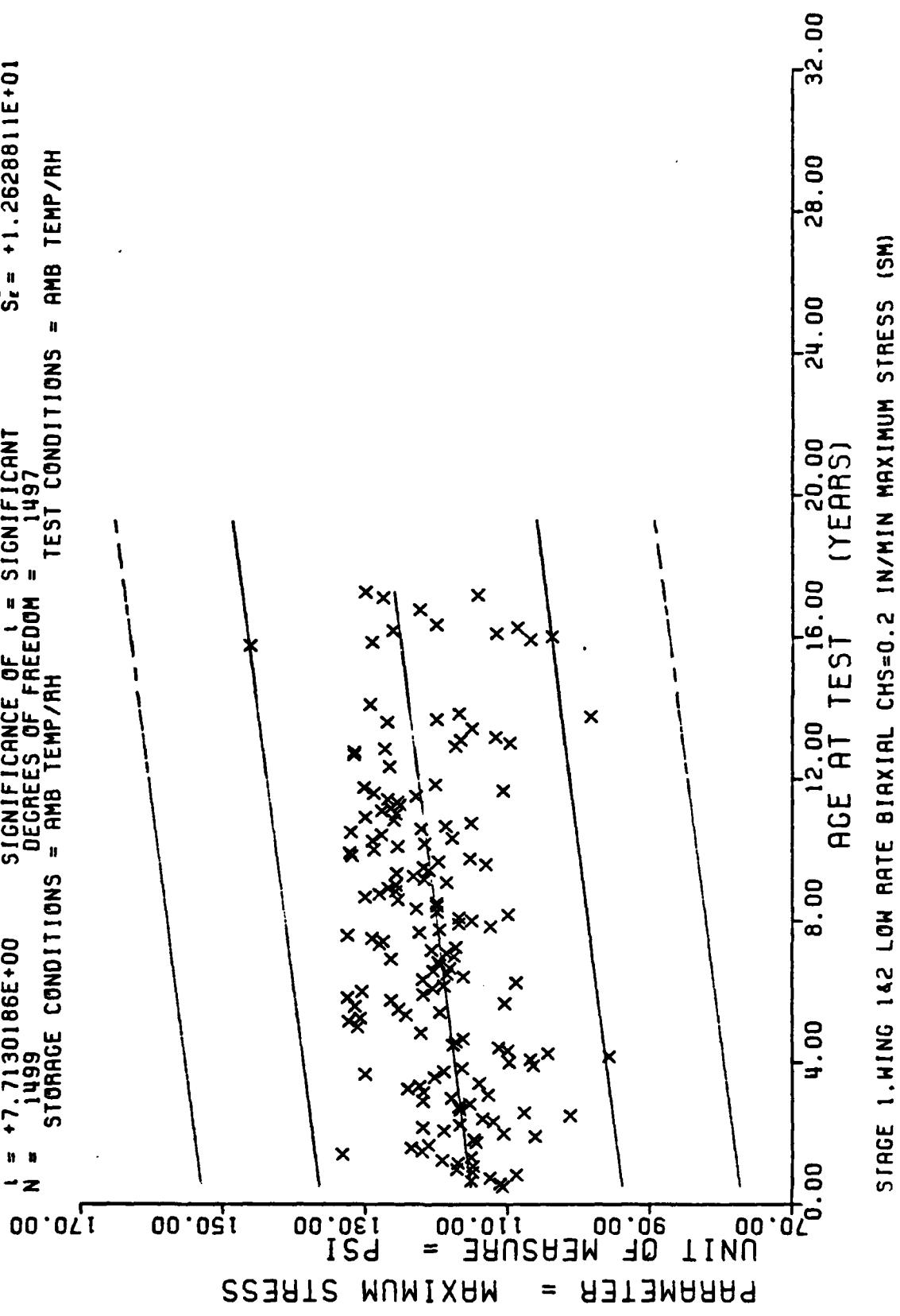


Figure 12

$\gamma = ((+2.3676610E-01) + (+2.1284582E-05) \times x)$   
 $F = +2.0037871E+00$  SIGNIFICANCE OF  $F = \text{NOT SIGNIFICANT}$   
 $R = +3.6598176E-02$  SIGNIFICANCE OF  $R = \text{NOT SIGNIFICANT}$   
 $I = +1.4155518E+00$  SIGNIFICANCE OF  $I = \text{NOT SIGNIFICANT}$   
 $N = 1496$  DEGREES OF FREEDOM = 1494  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

PARAMETER = STRAIN AT RUPTURE  
 UNIT OF MEASURE = IN/IN  
 0.19 0.23 0.27 0.31 0.35

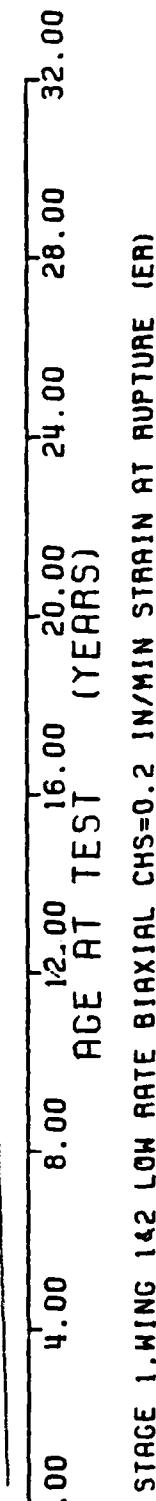


Figure 13

$F = +2.8009664E+01$        $( +1.1236573E+02 ) + ( +3.6091067E-02 ) * X$   
 $R = +1.3552445E-01$        $S_{\sigma} = +1.2536475E+01$   
 $t = +5.2924157E+00$        $S_{\epsilon} = +6.8193938E-03$   
 $N = 1499$        $S_{\epsilon\epsilon} = +1.2424961E+01$   
 DEGREES OF FREEDOM = 1497      TEST CONDITIONS = AMB. TEMP/RH  
 STORAGE CONDITIONS = AMB TEMP/RH

PARAMETER = STRESS AT RUPTURE

UNIT OF MEASURE = PSI

60.00 80.00 100.00 120.00 140.00 160.00

60.00 80.00 12.00 16.00 20.00 24.00 28.00 32.00

STAGE 1, WING 142 LOW RATE BIAXIAL CHS=0.2 IN/MIN STRESS AT RUPTURE (SR)

Figure 14

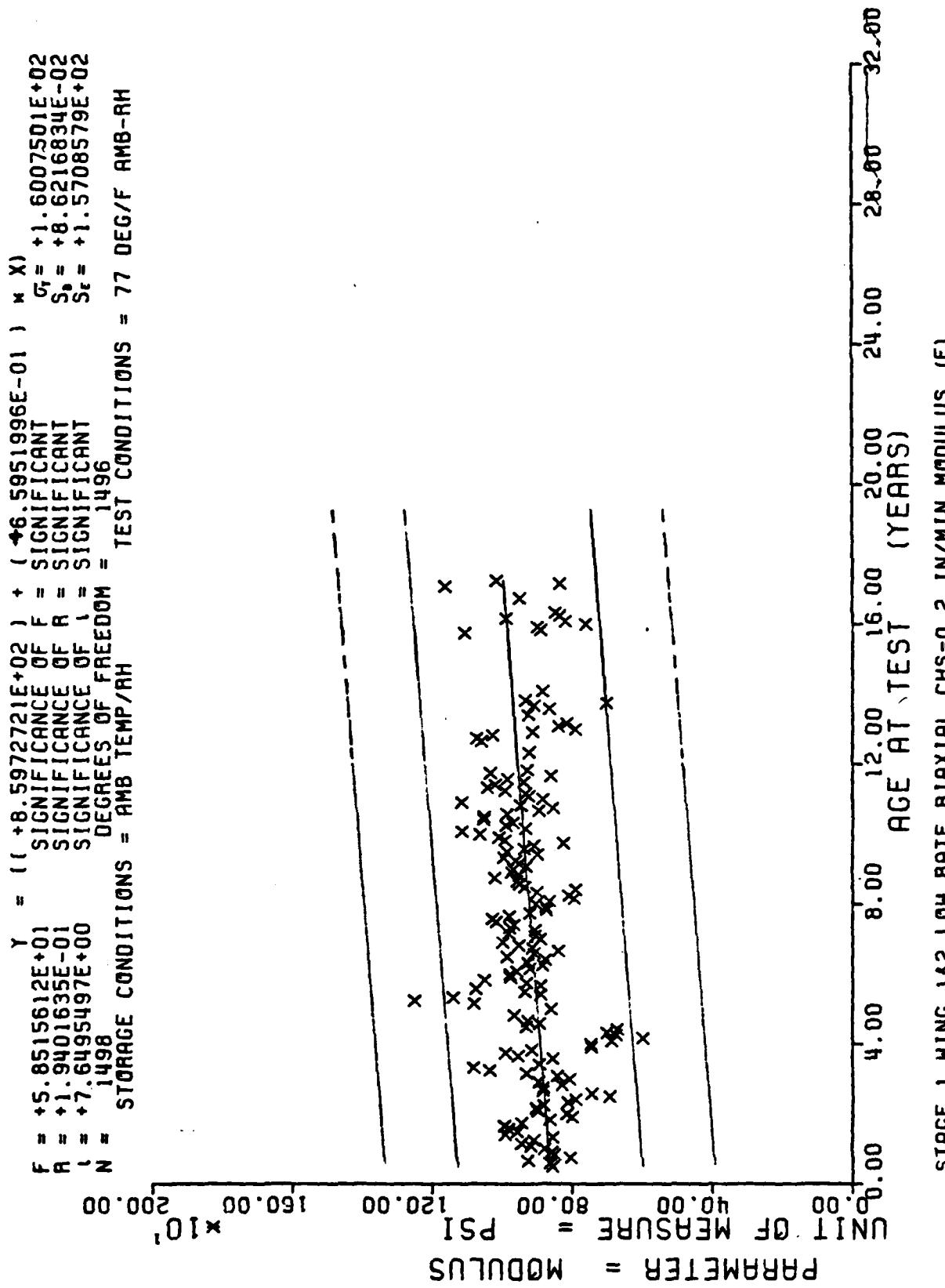


Figure 15

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP								
2	1	32	19	63	12	88	101	113	48
3	1	33	21	64	12	89	75	114	26
4	2	34	11	65	6	90	85	115	18
5	4	35	14	66	12	91	66	116	35
6	10	36	12	67	9	92	96	117	140
7	11	37	9	68	12	93	49	118	141
8	12	38	1	69	15	94	61	119	142
9	13	39	4	70	12	95	42	120	143
10	14	40	3	71	21	96	30	121	144
11	15	41	3	72	21	97	26	122	145
12	16	41	7	73	24	98	24	123	146
13	17	42	14	73	12	99	18	124	147
14	18	43	34	74	18	100	15	125	148
15	19	44	31	75	9	101	18	126	149
16	20	45	29	76	27	102	18	127	150
17	21	46	32	77	45	103	20	128	151
18	22	47	30	78	18	104	42	129	152
19	23	48	6	79	36	105	27	130	153
20	24	49	10	80	27	106	31	131	154
21	25	50	8	81	21	107	18	132	155
22	26	51	5	82	63	108	21	133	156
23	27	52	3	83	24	109	23	134	157
24	28	53	84	84	39	110	30	135	158
25	29	54	3	85	22	111	52	136	159
26	30	55	6	86	20	112	20	137	160
27	31	56	9	87	69	113	33	138	161

STAGE 1. WING 162 HIGH RATE

This sample size summary is applicable to figures 16 thru 20

CHS=1750 IN/MIN STRAIN AT MAX STRESS (EN) 194

193

195

196

197

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199

200

201

202

203

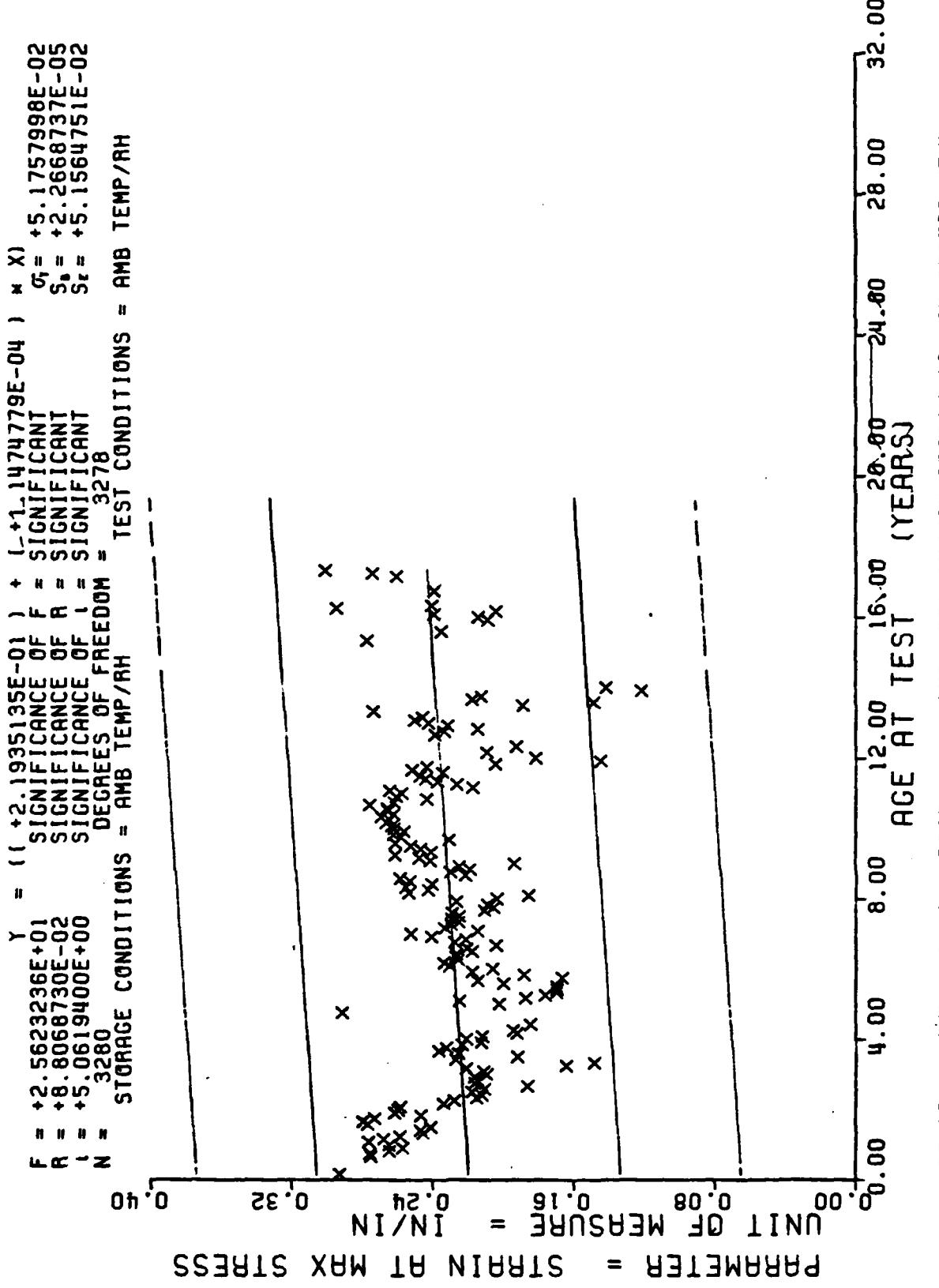
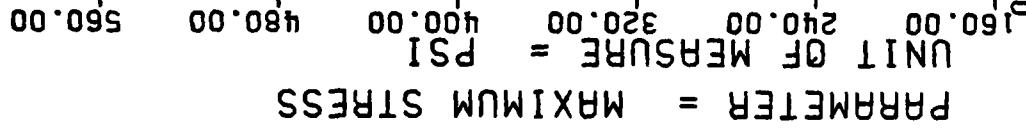


Figure 16

$F = +2.5424648E-01$     $Y = ( ( +3.7381656E+02 ) + ( +1.0465506E-02 ) * X )$   
 $R = +8.8065572E-03$     $\sigma_f = +4.7207279E+01$     $S_f = +2.0755479E-02$   
 $I = +5.0422860E-01$     $S_a = +4.7212648E+01$   
 $N = 3280$     $S_r = +4.7212648E+01$   
DEGREES OF FREEDOM = 3278   TEST CONDITIONS = AMB TEMP/RH

STORAGE CONDITIONS = AMB TEMP/RH   TEST CONDITIONS = AMB TEMP/RH



STAGE 1, WING 142 HIGH RATE   CHS=1750 IN/MIN MAXIMUM STRESS (SM)

Figure 17

$y = (1 + 3.5694579E-01) + ( - 2.2871608E-04) \times x$   
 $F = +2.0210080E+02$   
 $F = \text{SIGNIFICANCE OF } F$   
 $R = -2.4101857E-01$   
 $R = \text{SIGNIFICANCE OF } R$   
 $t = +1.4216216E+01$   
 $t = \text{SIGNIFICANCE OF } t$   
 $l = 3279$   
 $l = \text{DEGREES OF FREEDOM} = 3277$   
 $N = \text{STORAGE CONDITIONS = AMB TEMP/RH}$

TEST CONDITIONS = AMB TEMP/RH

PARAMETER = STRAIN AT RUPTURE

UNIT OF MEASURE = IN/IN

AGE AT TEST (YEARS) CH5=1750 IN/MIN STRAIN AT RUPTURE (ERI)

0.00 4.00 8.00 12.00 16.00 20.00 24.00 28.00 32.00

Figure 18

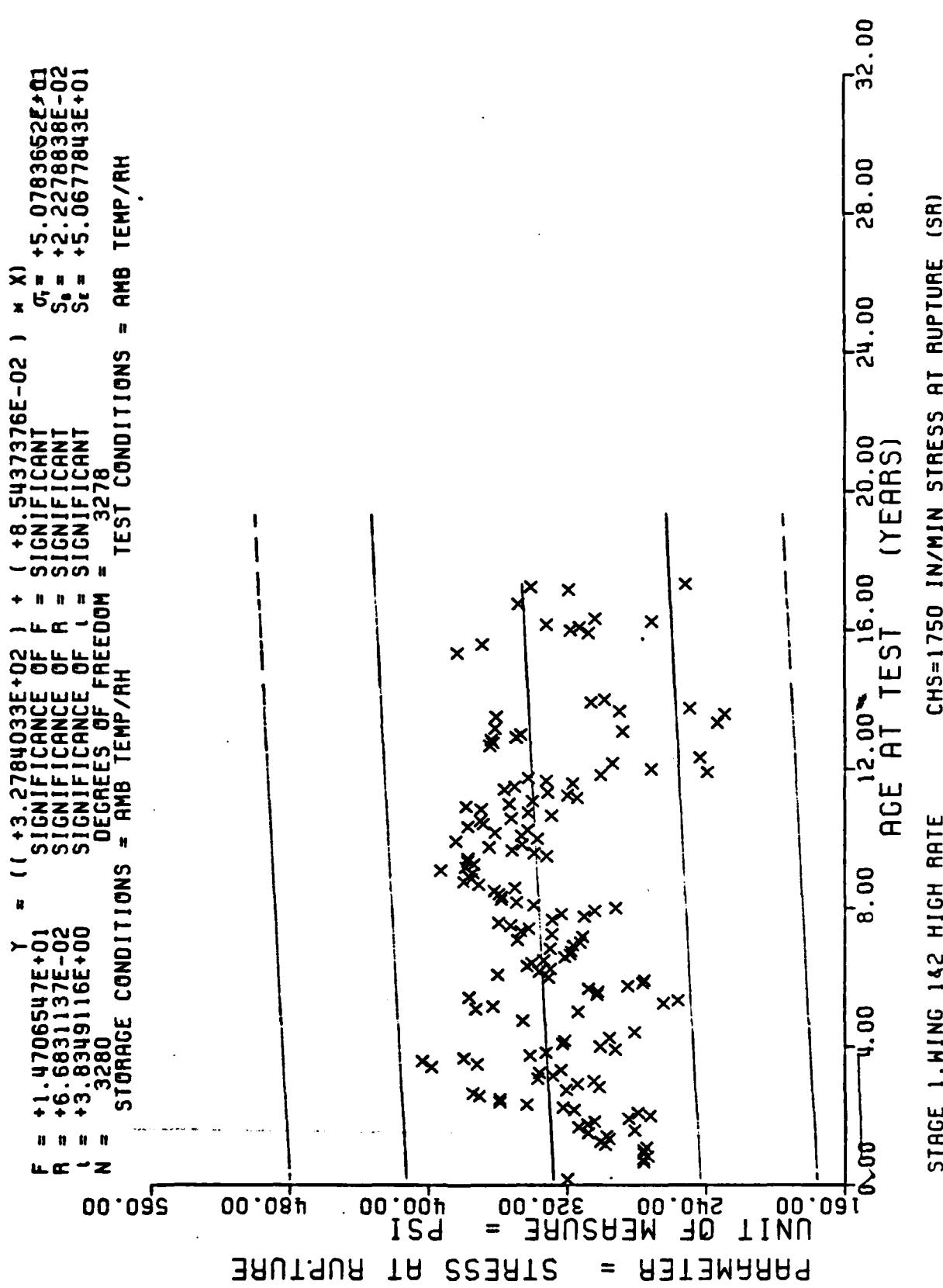


Figure 19

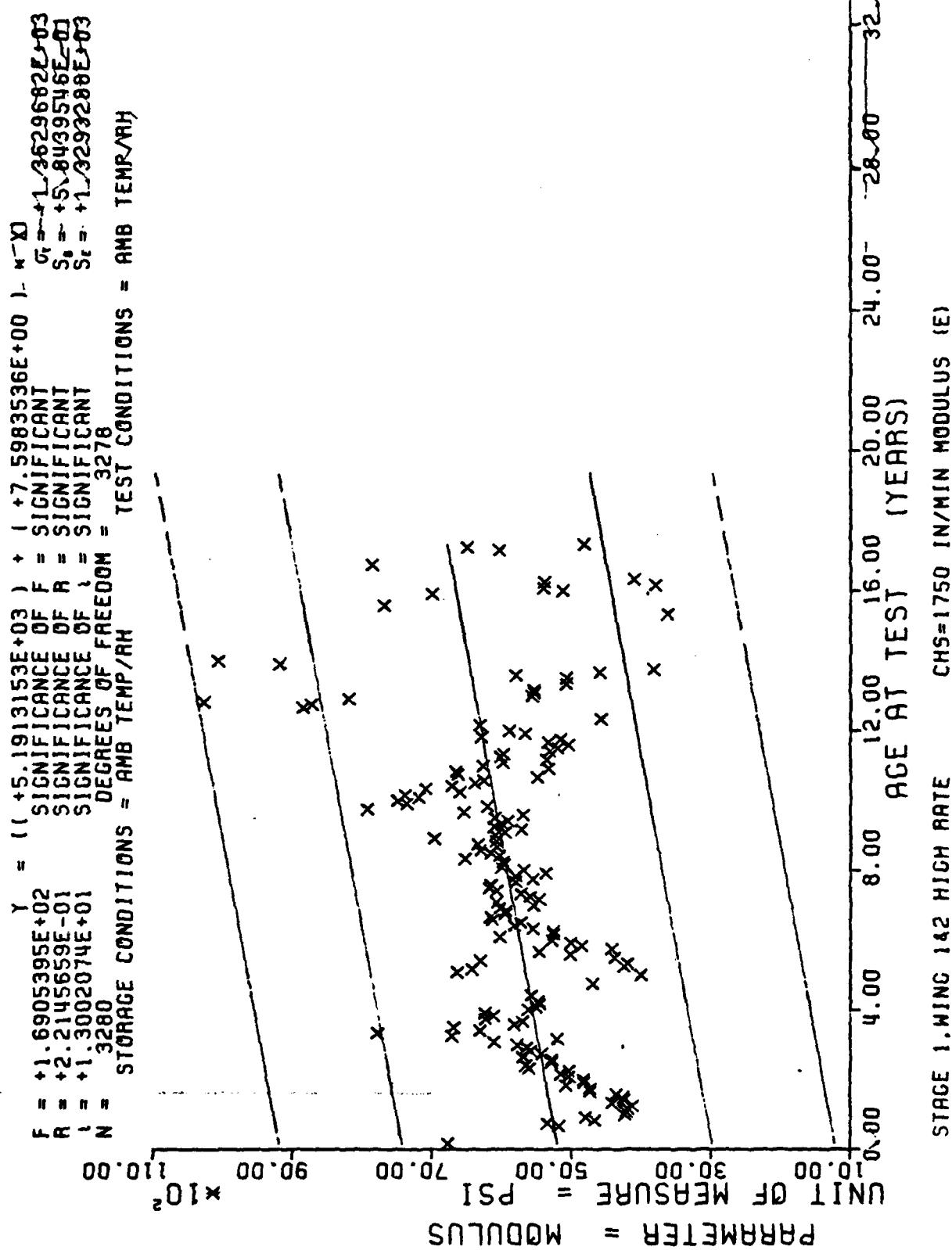


Figure 20

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (WCS)	NR SAMP	AGE (WOS)	NR SAMP	AGE (WOS)	NR SAMP	AGE (WCS)	NR SAMP	AGE (WCS)	NR SAMP	AGE (WOS)	NR SAMP	AGE (WOS)	NR SAMP	AGE (WOS)	
22	1	60	3	87	14	112	14	137	6	171	6	188	2	188	2
23	1	63	4	88	21	113	7	138	3	189	2	189	2	189	2
25	2	64	4	89	25	114	5	139	5	191	6	191	6	191	6
31	1	65	3	90	28	115	6	140	3	192	2	192	2	192	2
32	1	66	3	91	13	116	5	141	3	193	8	193	8	193	8
33	2	67	2	92	5	117	8	142	2	194	2	194	2	194	2
34	2	68	2	93	6	118	9	143	5	195	3	195	3	195	3
35	1	69	2	94	8	119	11	144	10	196	1	196	1	196	1
36	3	70	3	95	9	120	11	145	1	197	1	197	1	197	1
37	2	71	6	96	8	121	9	146	3	199	2	199	2	199	2
39	2	72	9	97	8	122	16	147	2	202	4	202	4	202	4
40	1	73	4	98	18	123	6	148	1	203	2	203	2	203	2
41	5	74	7	99	10	124	5	149	1	204	1	204	1	204	1
42	2	75	18	100	10	125	16	150	2	205	2	205	2	205	2
43	8	76	5	101	8	126	3	153	8	206	8	206	8	206	8
44	15	77	5	102	10	127	7	154	3	207	3	207	3	207	3
45	29	78	10	103	5	128	16	155	12						
46	25	79	5	104	7	129	8	156	9						
47	35	80	11	105	7	130	7	157	3						
48	7	81	4	106	12	131	3	160							
49	12	82	12	107	13	132	8	163	2						
50	4	83	7	108	14	133	12	164	1						
51	3	84	4	109	8	134	12	166							
53	2	85	15	110	12	135	2	168	4						
59	1	86	5	111	7	136	8	169	3						

$$Y = (( +2.1680176E-01 ) + ( +3.100038E-04 ) \ln X)$$

$$F = +9.7074043E+01 \quad \text{SIGNIFICANT}$$

$$F = +3.0063147E-01 \quad \text{SIGNIFICANT}$$

$$F = +9.8526160E+00 \quad \text{SIGNIFICANT}$$

$$F = +9.79 \quad \text{DEGREES OF FREEDOM} = 977$$

$$N = \text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$$

$$\sigma_f = +4.0819771E-02$$

$$S_f = +3.1565260E-05$$

$$S_f = +3.8951383E-02$$

$$\text{TEST CONDITIONS} = \text{AMB TEMP/RH}$$

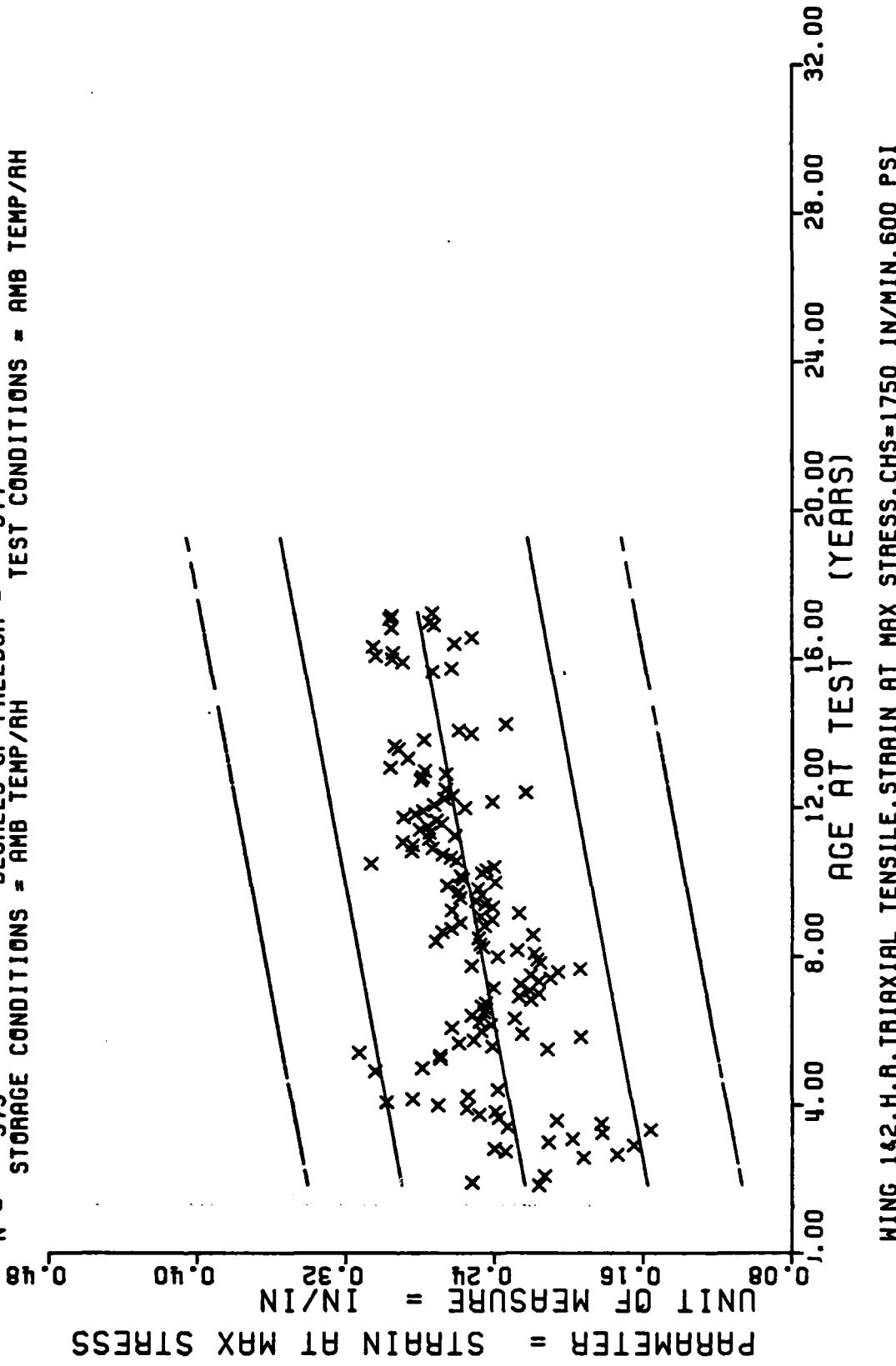
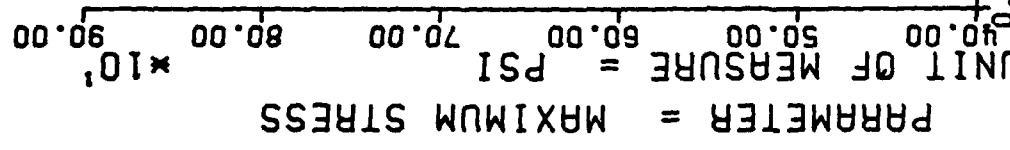


Figure 21

$F = +2.1911406E+01$        $\gamma = (( +6.0889499E+02 ) + ( +2.3518610E-01 ) \times X)$   
 $R = +1.4810565E-01$       SIGNIFICANCE OF  $F$  = SIGNIFICANT  
 $I = +4.6809621E+00$       SIGNIFICANCE OF  $R$  = SIGNIFICANT  
 $N = 979$       SIGNIFICANCE OF  $I$  = SIGNIFICANT  
 $D$ EGREES OF FREEDOM = 977      TEST CONDITIONS = AMB TEMP/RH  
 STORAGE CONDITIONS = AMB TEMP/RH



WING 142, H.R. TRIAXIAL TENSILE, MAXIMUM STRESS, CHS=1750 IN/MIN, 600 PSI

Figure 22

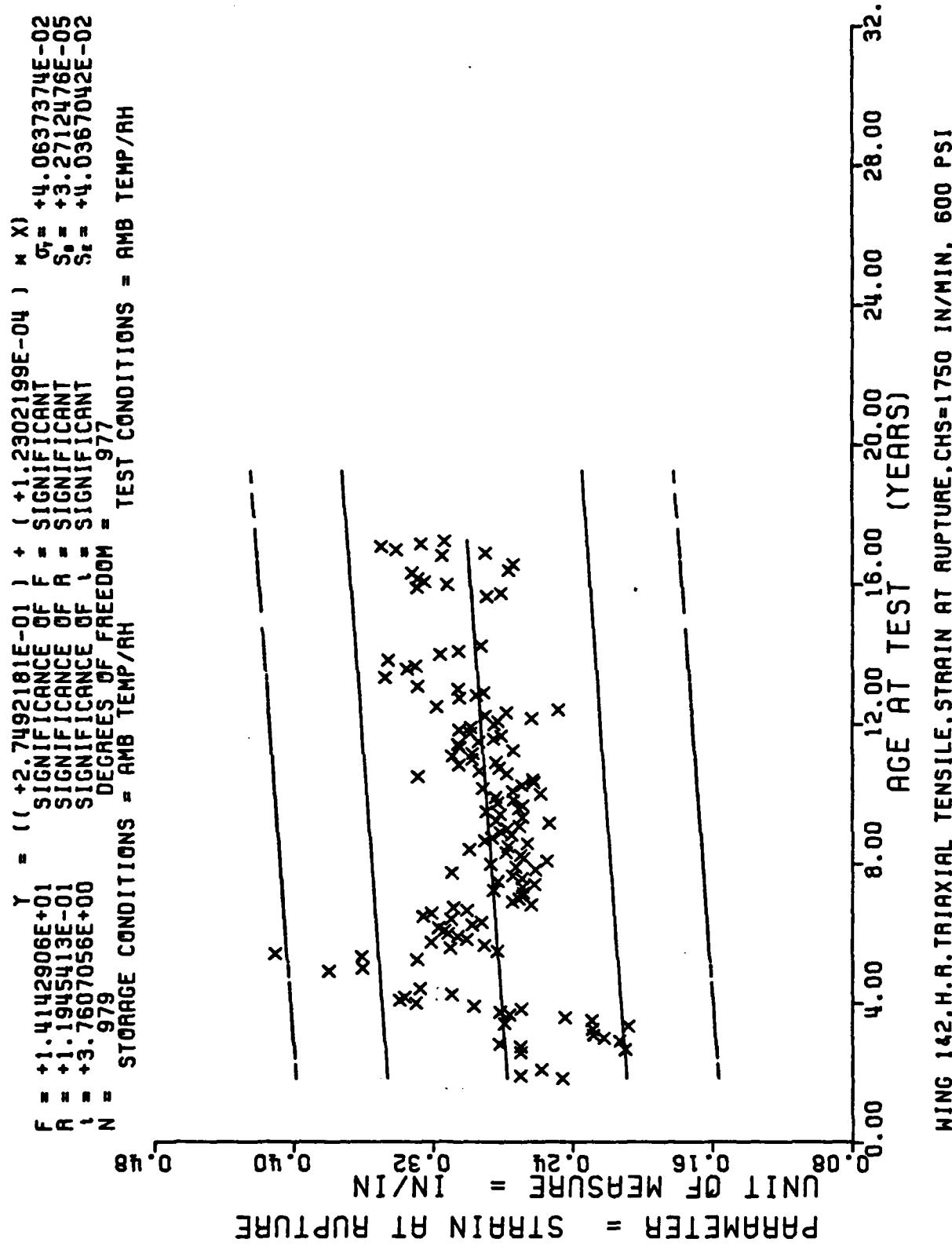


Figure 23

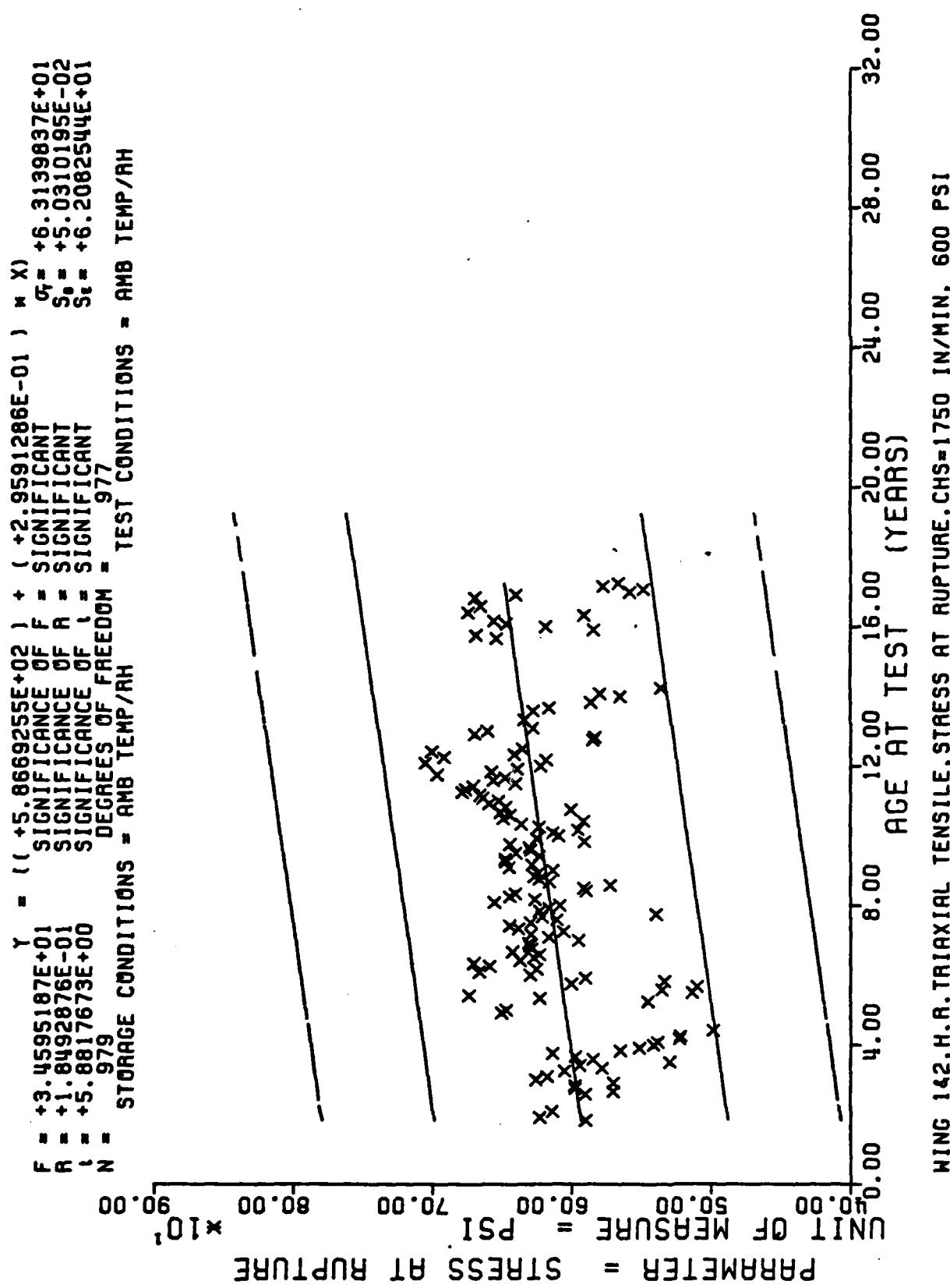
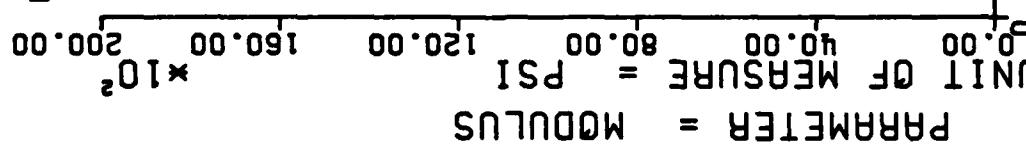


Figure 24

$Y = (( +6.7596788E+03) + (-5.3877671E+00) \times X)$   
 $F = \text{SIGNIFICANCE OF } F = \text{SIGNIFICANT}$   
 $R = \text{SIGNIFICANCE OF } R = \text{SIGNIFICANT}$   
 $I = \text{SIGNIFICANCE OF } I = \text{SIGNIFICANT}$   
 $N = 3.9278922E+00 \quad \text{DEGREES OF FREEDOM} = 977$   
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH} \quad \text{TEST CONDITIONS} = \text{AMB TEMP/RH}$



WING 142, H.R. TRIAXIAL TENSILE MODULUS, CHS=1750 IN/MIN AT 600 PSI

Figure 25

\*\*\*\* SAMPLE SIZE SUMMARY \*\*\*\*

AGE (MOS)	NR SAMP								
7	4	32	39	64	69	114	114	139	27
8	3	33	38	65	90	115	115	140	18
9	5	34	30	66	91	113	116	141	33
10	4	35	29	67	92	117	117	142	15
11	5	36	29	68	93	118	21	144	3
12	7	37	23	69	94	119	36	145	3
13	6	38	20	70	95	120	42	147	6
14	10	39	12	71	96	121	15	151	3
15	12	40	28	72	97	122	9	152	3
16	9	41	38	73	98	123	12	153	24
17	6	42	84	74	21	99	15	154	15
18	8	43	63	75	21	100	48	125	24
19	7	44	50	76	30	101	36	126	44
20	11	45	39	77	15	102	27	127	51
21	12	46	48	78	27	103	43	128	21
22	9	47	36	79	27	104	34	129	33
23	8	48	15	80	30	105	27	130	18
24	6	49	17	81	18	106	33	131	27
25	18	50	5	82	36	107	24	132	28
26	13	51	4	83	27	108	33	133	18
27	17	52	3	84	27	109	27	134	33
28	17	53	5	85	33	110	21	135	36
29	29	58	3	86	34	111	18	136	39
30	29	62	3	87	69	112	36	137	33
31	51	63	3	88	69	113	21	138	15
								192	15
								193	12
								194	7
								196	6
								197	3
								202	3
								203	6
								204	9
								206	18

STAGE 1, WING 162, STRESS RELAXATION J-0X 10 SEC TEST TEMP. 77 DEG

This sample size summary is applicable to figures 26 thru 29

$F = +3.5067171E+02$        $F = (+4.6858771E+02)$        $F = (+8.3850018E-01)$        $F = (+1.1603648E+02)$   
 $R = +2.9463024E-01$        $R = SIGNIFICANT$        $S_r = +4.4776773E-02$   
 $L = +1.8726230E+01$        $R = SIGNIFICANT$        $S_r = +1.1090080E+02$   
 $N = 3689$        $L = SIGNIFICANT$        $S_r = +1.1090080E+02$   
 $DEGREES OF FREEDOM = 3689$        $TEST CONDITIONS = +077 DEG/RH$

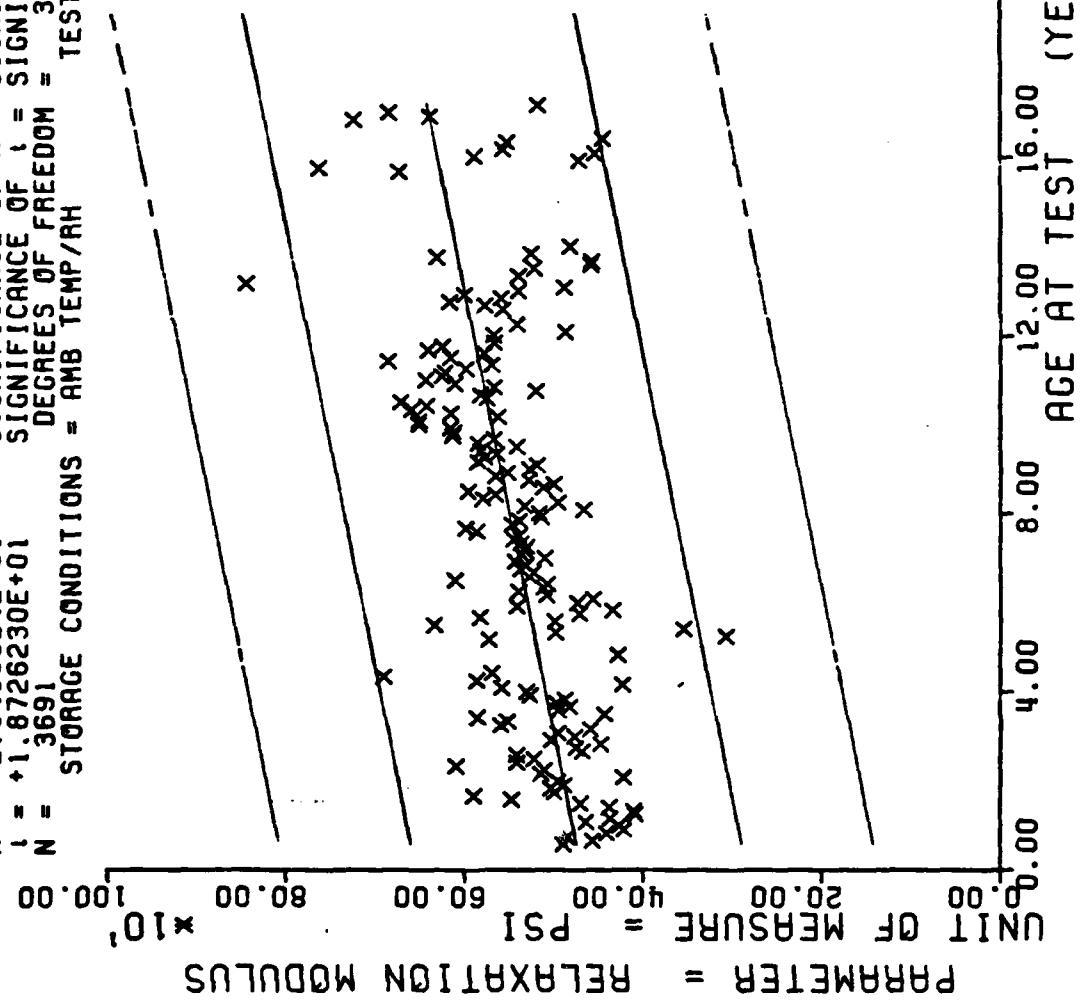
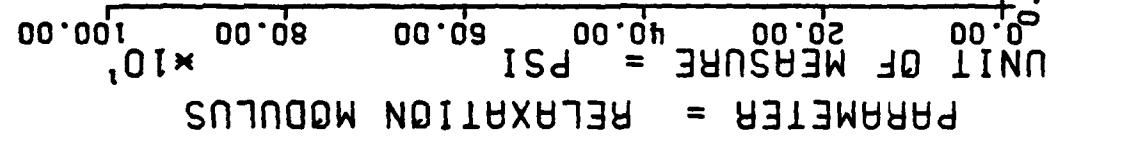


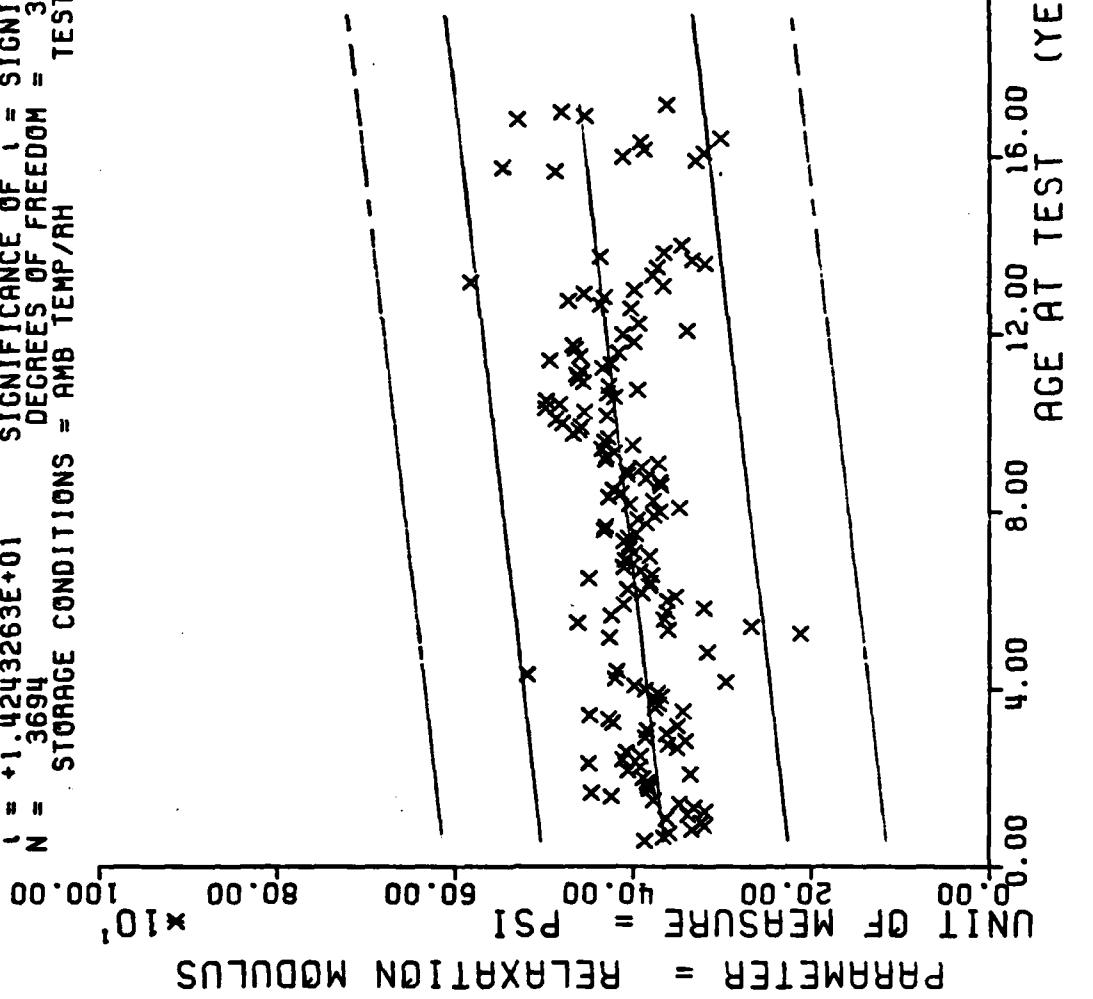
Figure 26

$\gamma = ( ( +3.862654 \times 10^02 ) + ( +5.2623056 \times 10^01 ) \times X )$   
 $F = +2.1642167 \times 10^02$  SIGNIFICANCE OF  $F$  = SIGNIFICANT  
 $R = +2.3531502 \times 10^01$  SIGNIFICANCE OF  $R$  = SIGNIFICANT  
 $t = +1.4711276 \times 10^01$  SIGNIFICANCE OF  $t$  = SIGNIFICANT  
 $N = 3694$  DEGREES OF FREEDOM = 3692 TEST CONDITIONS = +077 DEG/RH  
 STORAGE CONDITIONS = AMB TEMP/RH

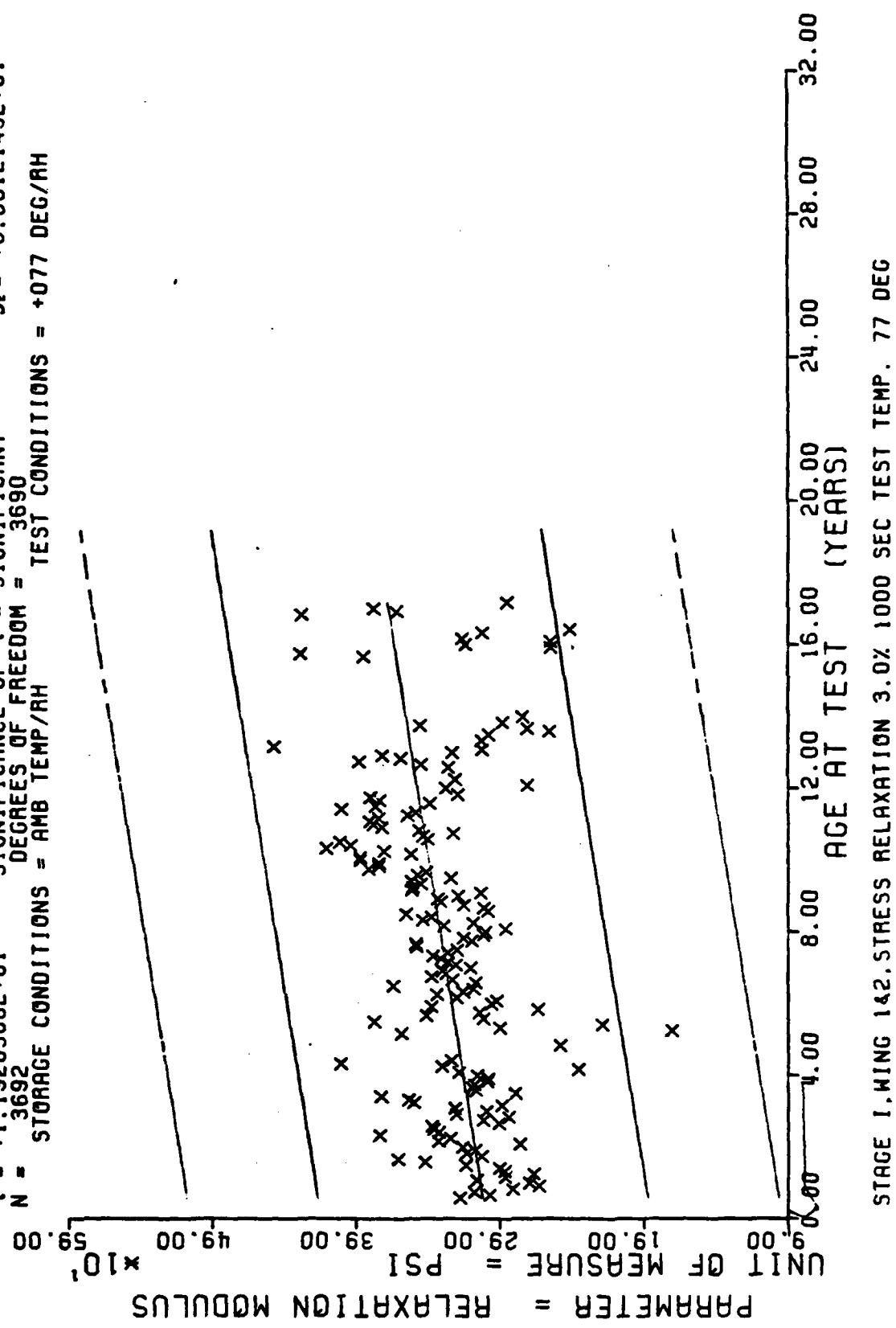


$F = +2.0287055E+02$   
 $R = +2.2822488E-01$   
 $S = +1.4243263E+01$   
 $N = 3694$   
 STORAGE CONDITIONS = AMB TEMP/RH

$\gamma = (( +3.6118774E+02 ) + ( +4.7783457E-01 ) \times X)$   
 SIGNIFICANCE OF  $F$  = SIGNIFICANT  
 SIGNIFICANCE OF  $R$  = SIGNIFICANT  
 SIGNIFICANCE OF  $S$  = SIGNIFICANT  
 DEGREES OF FREEDOM = 3692  
 TEST CONDITIONS = +077 DEG/RH



$F = +1.4209567E+02$  SIGNIFICANCE OF  $F$  = SIGNIFICANT  
 $R = +1.9256273E-01$  SIGNIFICANCE OF  $R$  = SIGNIFICANT  
 $A = +1.1920388E+01$  SIGNIFICANCE OF  $A$  = SIGNIFICANT  
 $N = 3692$  DEGREES OF FREEDOM = 3690  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = +077 DEG/RH



## ◆◆◆ SAMPLE SIZE SUMMARY ◆◆◆

STAGE 1. WING 162. STRESS RELAXATION 5.0X 10 SEC TEST TEMP. 77 DEG

This sample size summary is applicable to figures 30 thru 33

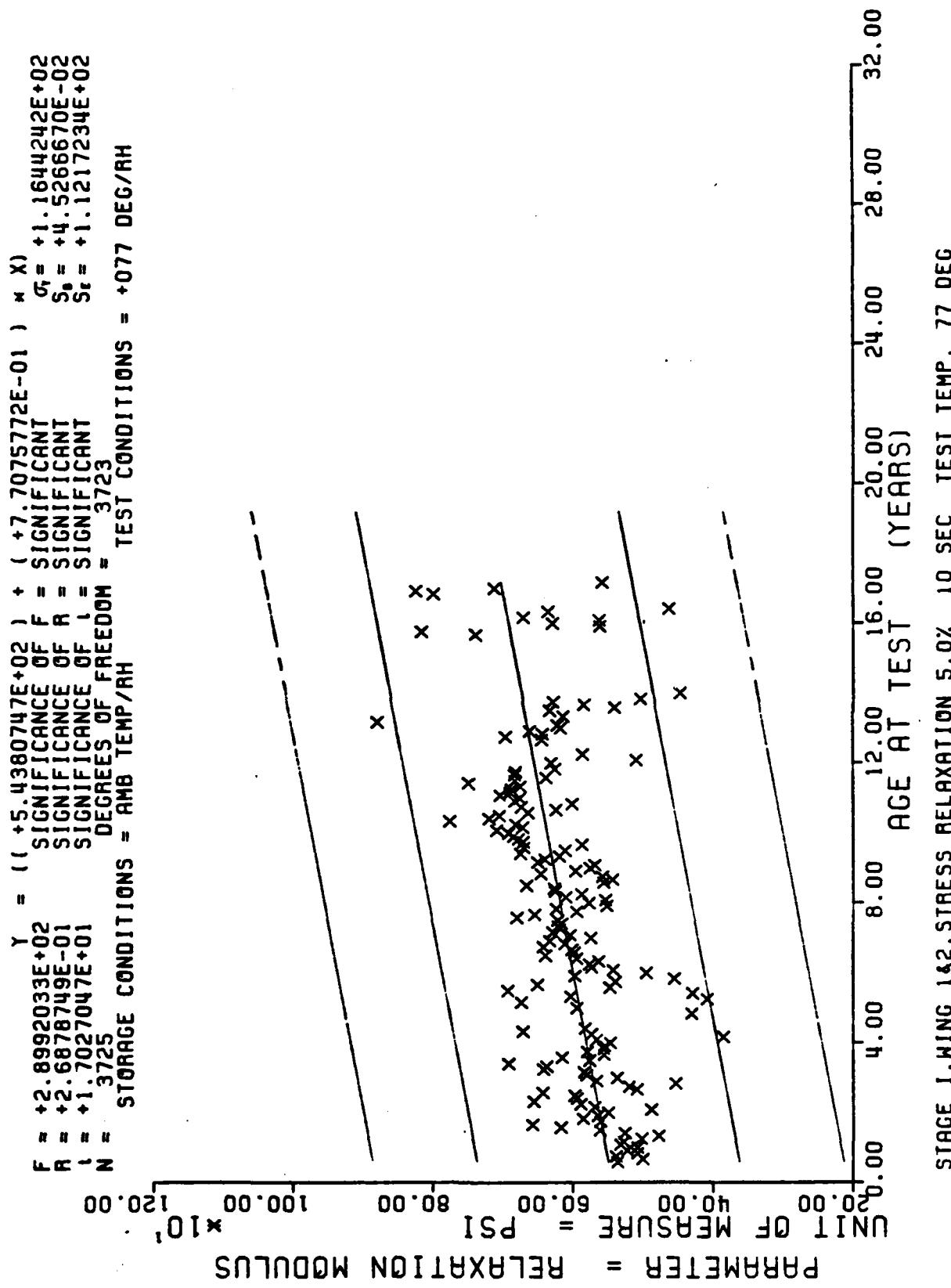


Figure 30

$F = +1.0178454E+02$        $\gamma = (( +4.3405452E+02 ) + ( +3.4163155E-01 ) * X)$   
 $F = \text{SIGNIFICANCE OF } F$        $\sigma_t = +8.5039946E+01$   
 $R = +1.6313135E-01$        $F = \text{SIGNIFICANCE OF } R$        $S_t = +3.3862347E-02$   
 $R = +1.0088832E+01$        $R = \text{SIGNIFICANCE OF } R$        $S_c = +8.3912049E+01$   
 $N = 3725$        $\text{DEGREES OF FREEDOM} = 3723$        $\text{TEST CONDITIONS} = +077 \text{ DEG/RH}$

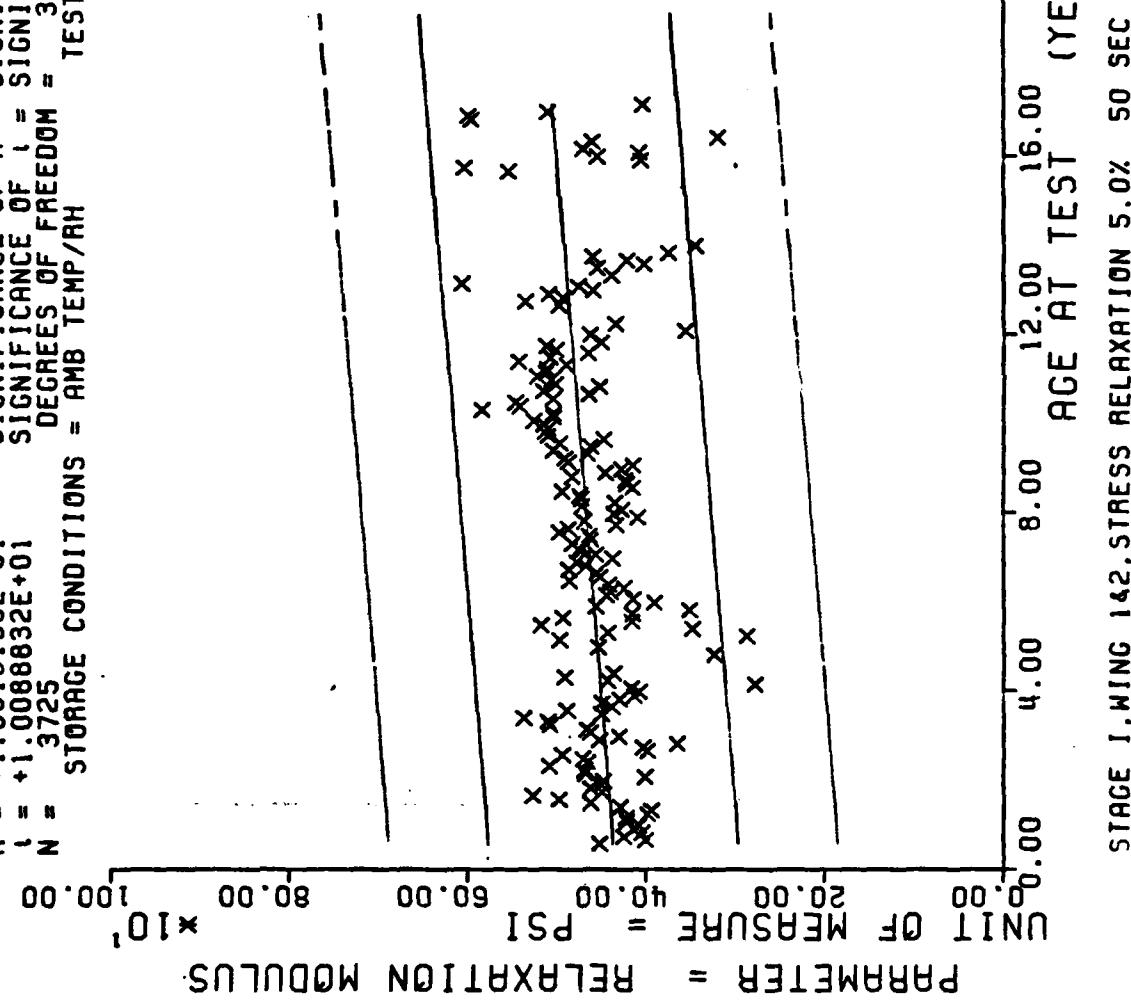


Figure 31

$F = +8.6874458E+01$     $\gamma = 11.0584774E+02$     $\beta = (+2.9375695E-01)$     $\alpha = (+2.9375695E-01)$   
 $F =$  SIGNIFICANT    $\sigma = +7.8995046E+01$   
 $\alpha =$  SIGNIFICANT    $S_a = +3.1516798E-02$   
 $\beta =$  SIGNIFICANT    $S_c = +7.8099699E+01$   
 $\gamma =$  SIGNIFICANT  
DEGREES OF FREEDOM = 3723   TEST CONDITIONS = +077 DEG/RH  
N = STORAGE CONDITIONS = AMB TEMP/RH

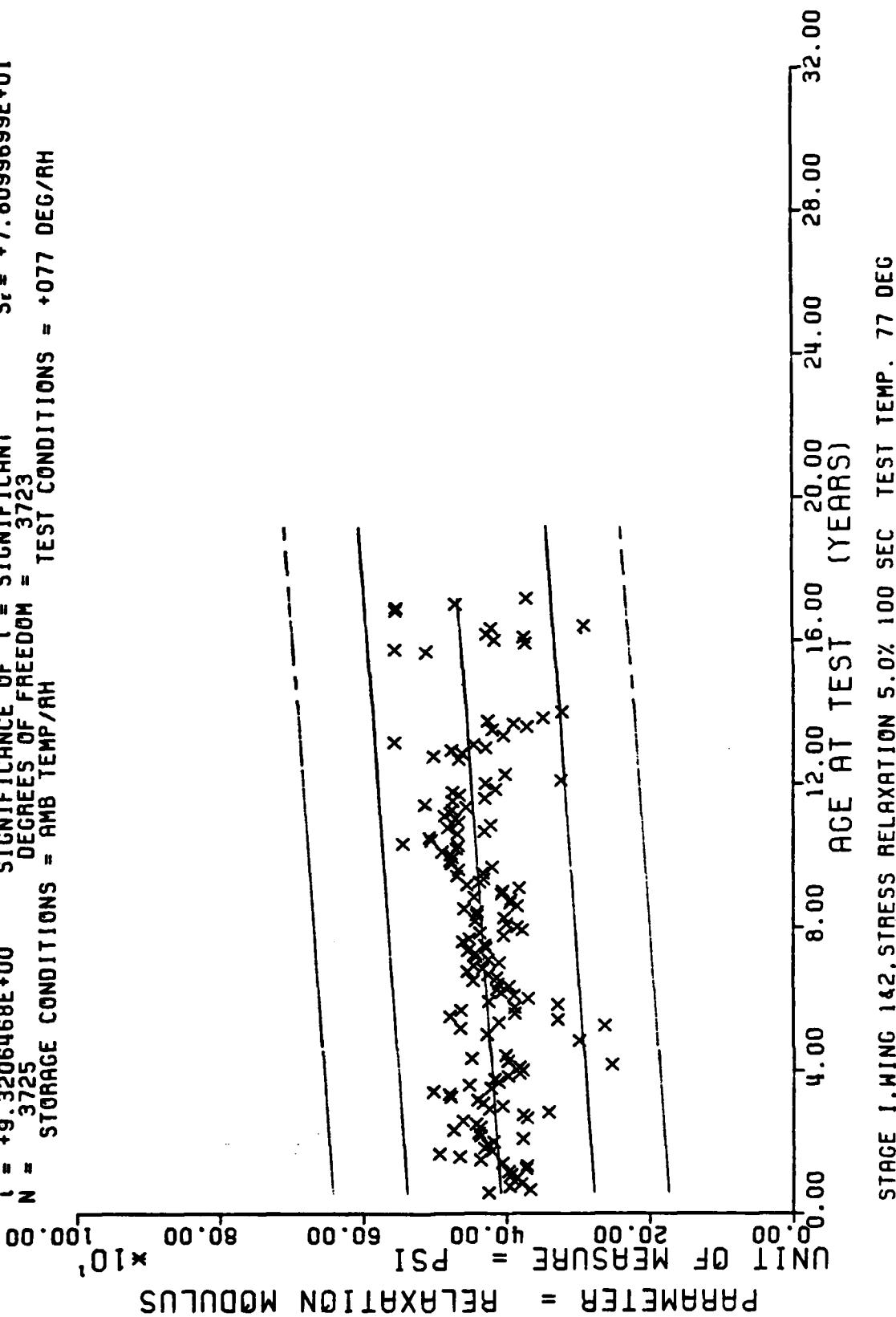


Figure 32

$F = +4.5826662E+01$        $\gamma = (( +3.3428323E+02 ) + i +1.7670769E-01) \times X)$   
 $R = +1.1035753E-01$       SIGNIFICANCE OF F = SIGNIFICANT  
 $t = +6.7695393E+00$       SIGNIFICANCE OF R = SIGNIFICANT  
 $N = 3719$       SIGNIFICANCE OF I = SIGNIFICANT  
DEGREES OF FREEDOM = 3717  
STORAGE CONDITIONS = AMB TEMP/RH      TEST CONDITIONS = +077 DEG/RH

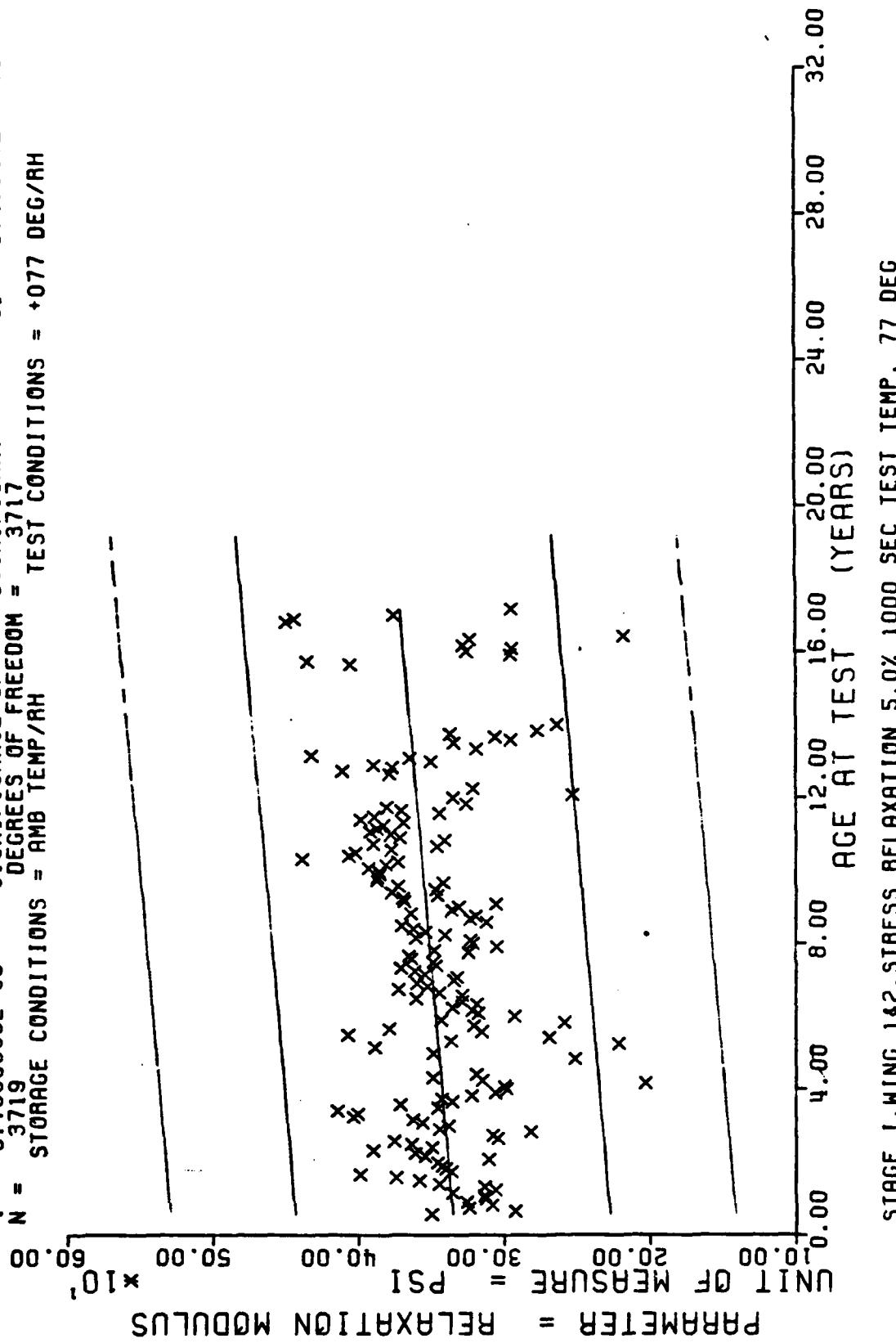
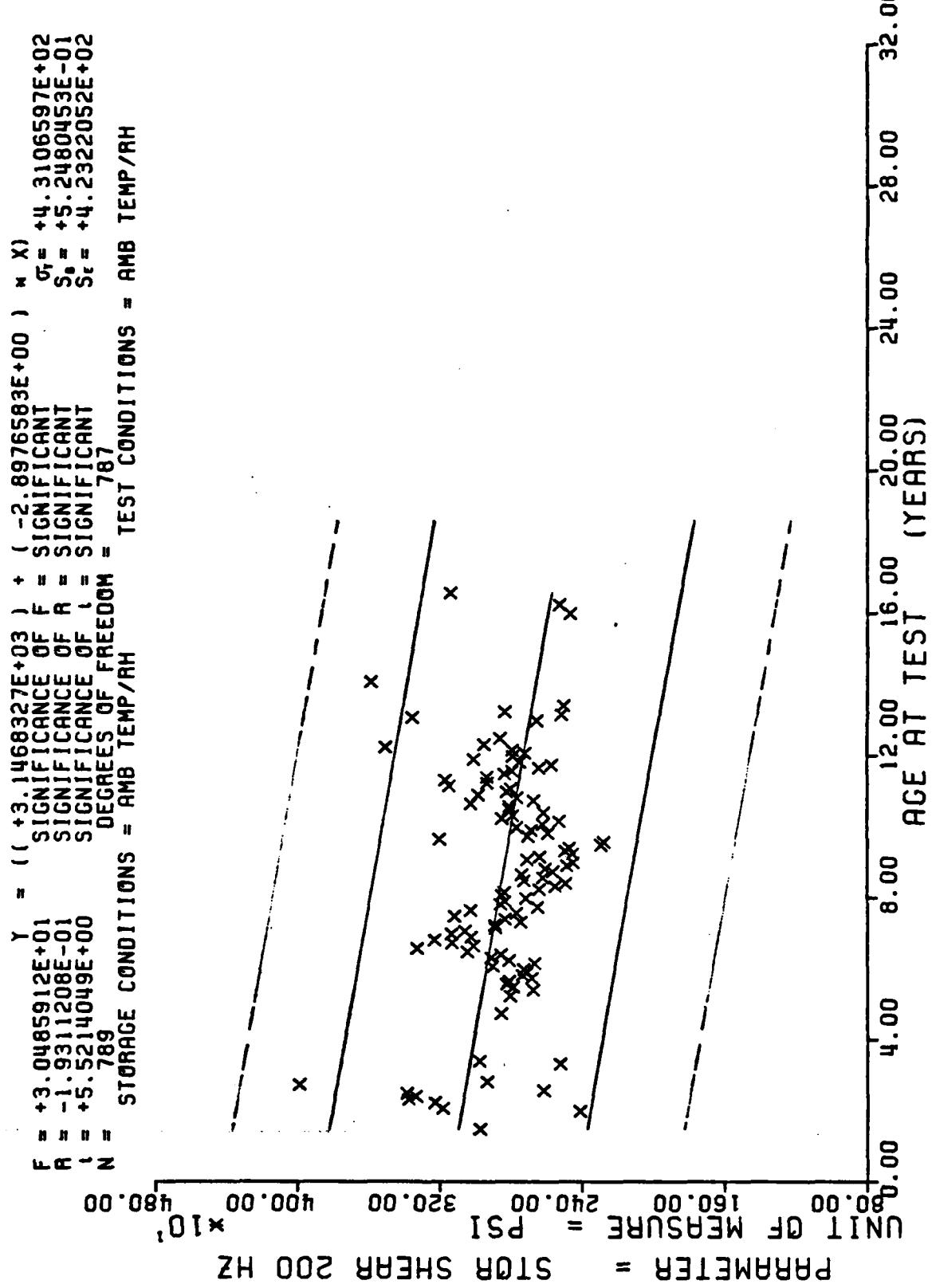


Figure 33

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (NCS)	NR SAMP								
18	1	76	6	101	7	126	7	157	1
24	1	77	9	102	7	127	11	158	1
25	3	78	5	103	11	128	5	159	1
27	7	79	5	104	6	129	6	161	3
28	8	80	5	105	6	130	11	169	1
29	8	81	9	106	13	131	10	192	1
30	11	82	5	107	3	132	18	195	1
31	3	83	11	108	6	133	7	199	1
32	2	84	8	109	11	134	7		
33	3	85	6	110	12	135	9		
34	1	86	9	111	8	136	13		
40	1	87	14	112	4	137	11		
41	1	88	14	113	3	138	5		
57	1	89	13	114	6	139	5		
63	3	90	45	115	3	140	6		
65	3	91	33	116	20	141	6		
66	2	92	28	117	13	142	6		
67	4	93	23	118	3	143	5		
68	3	94	10	119	7	144	2		
69	4	95	7	120	11	145	2		
70	7	96	7	121	11	146	2		
71	3	97	9	122	6	147	2		
72	7	98	4	123	10	148	2		
73	7	99	4	124	9	150	1		
74	3	100	9	125	4	156	1		
75	7								



WING 142 S1 TP-H1011 DYNAMIC RESPONSE, CENTER-WT 70 GM. STOR SHEAR AT 200 Hz

Figure 34

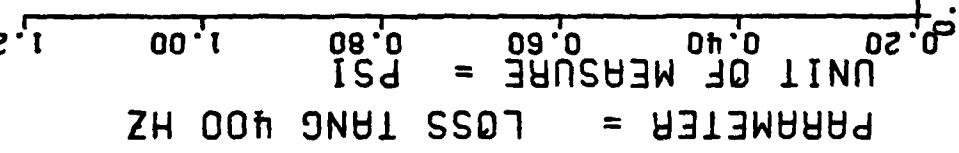
\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS.)	NR SAMP								
18	1	76	9	101	7	126	9	152	2
24	1	77	9	102	7	127	13	153	2
25	3	78	5	103	11	128	5	156	1
27	7	79	5	104	6	129	8	157	1
28	8	80	5	105	6	130	11	158	1
29	6	81	9	106	13	131	10	159	3
30	11	82	9	107	3	132	18	161	3
31	3	83	13	108	8	133	9	169	1
33	2	84	8	109	11	134	7	170	2
34	3	85	6	110	12	135	11	175	1
40	1	86	9	111	10	136	13	176	1
41	1	87	14	112	4	137	13	178	2
57	1	88	14	113	4	138	9	180	1
63	3	89	13	114	6	139	5	192	1
65	3	90	45	115	3	140	6	195	1
66	2	91	33	116	20	141	8	199	1
67	4	92	28	117	13	142	6	202	2
68	3	93	24	118	8	143	5		
69	4	94	10	119	7	144	2		
70	7	95	7	120	11	145	2		
71	6	96	7	121	11	146	2		
72	7	97	9	122	6	147	2		
73	7	98	4	123	10	148	2		
74	3	99	4	124	9	150	1		
	7	100	9	125	151		2		

WING 162 SI TP-H1011 DYNAMIC RESPONSE CENTER-WT 70 GM. STOR SHEAR AT 400 HZ

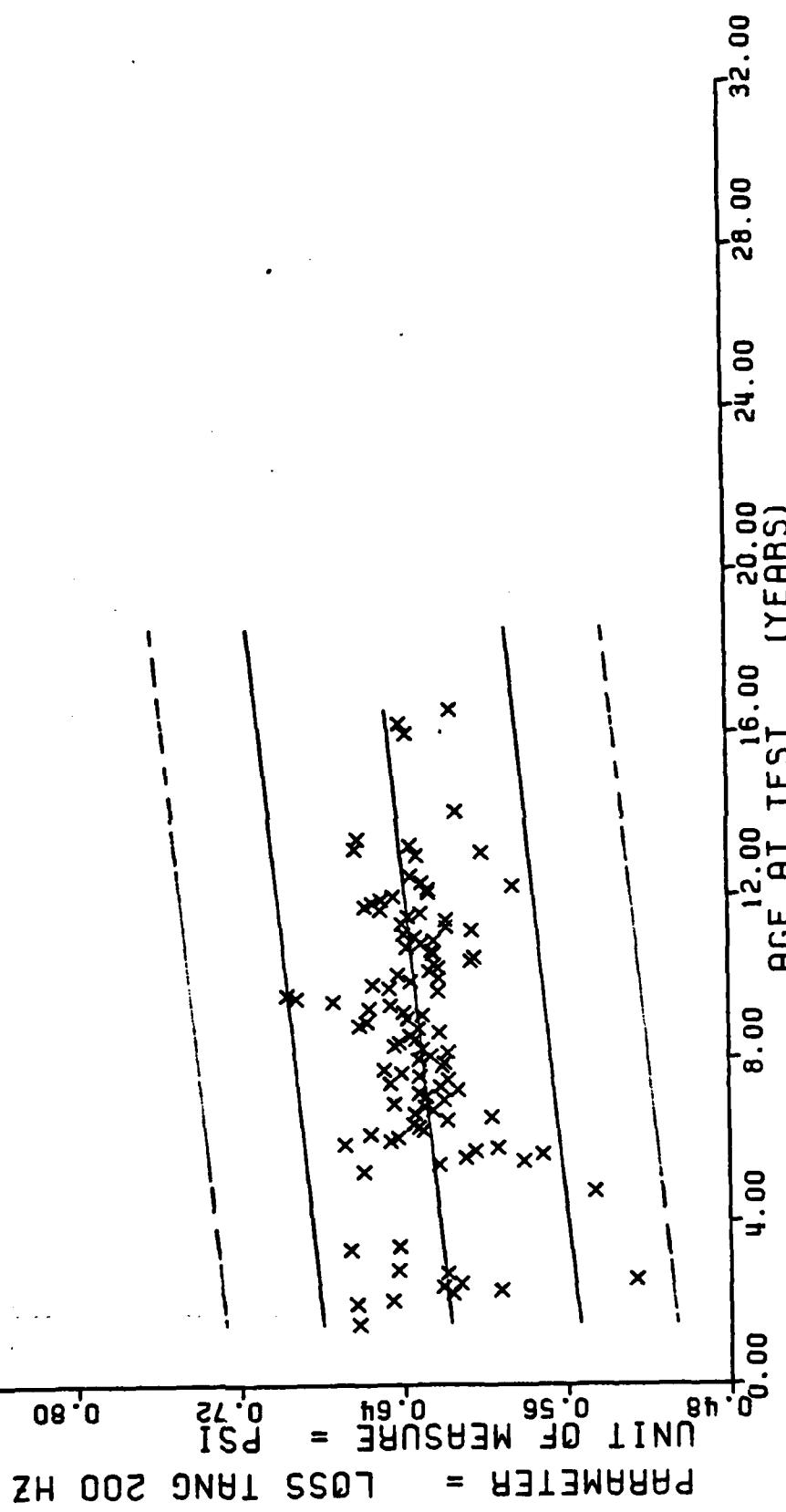
This sample size summary is applicable to figure 35, 36 and 37

$F = +2.7184294E+02$        $\gamma = (( +5.4605495E-01 ) + ( +1.2584997E-03 ) \times X)$   
 $R = +5.0669108E-01$       SIGNIFICANT  
 $I = +1.6487660E+01$       SIGNIFICANT  
 $N = 789$       SIGNIFICANT  
 $D$  = DEGREES OF FREEDOM = 787  
 $S$  = STORAGE CONDITIONS = AMB TEMP/RH      TEST CONDITIONS = AMB TEMP/RH



S1 WING 142 TP-H1011 DYNAMIC RESPONSE LOSS TANGENT AT 400 Hz, CENTER-WT 70 GM

$F = +1.3306961E+01$        $\gamma = (( +6.1385673E-01 ) + ( +1.6654159E-04 ) * X)$   
 SIGNIFICANCE OF  $F$  = SIGNIFICANT       $\sigma_F = +3.7103731E-02$   
 $R = +1.2894697E-01$       SIGNIFICANT       $S_0 = +4.5654465E-05$   
 $t = +3.6478708E+00$       SIGNIFICANT       $S_r = +3.6817339E-02$   
 DEGREES OF FREEDOM = 787  
 $N = 789$       TEST CONDITIONS = AMB TEMP/RH



S1 WING 142 1P-H1011 DYNAMIC RESPONSE LOSS TANGENT AT 200 Hz. CENTER-WT 70 GM

Figure 36

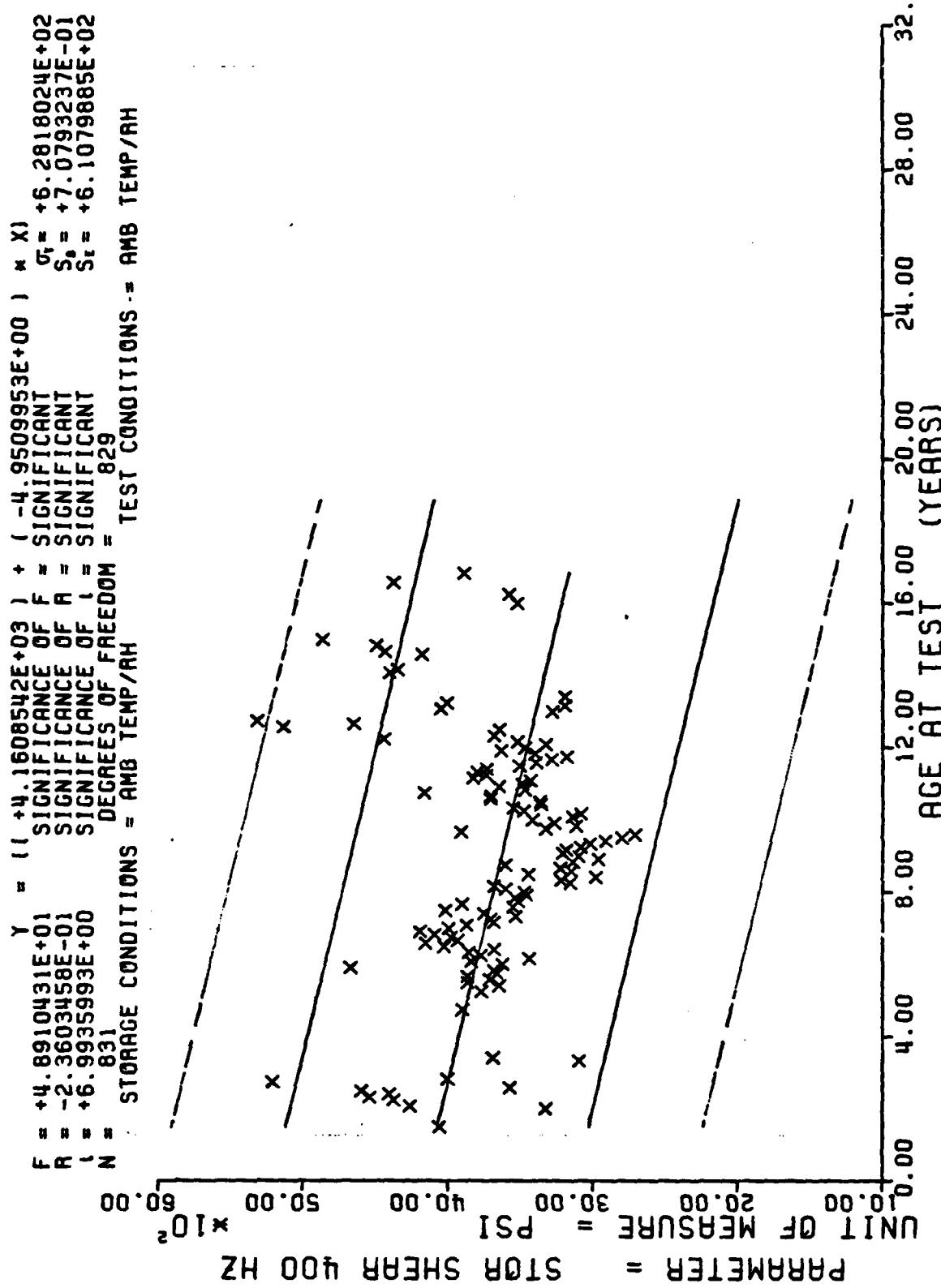


Figure 37

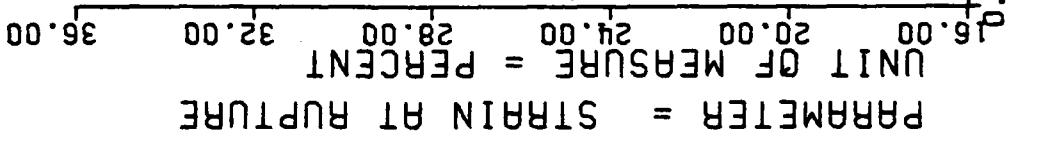
\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP								
2	1	31	24	56	1	88	15	113	30
7	1	32	18	64	2	89	24	114	15
8	3	33	14	65	1	90	11	115	16
9	3	34	9	66	3	91	5	116	21
10	5	35	9	67	3	92	32	117	15
11	3	36	6	68	4	93	25	118	18
12	8	37	9	69	2	94	39	119	33
13	5	38	3	70	6	95	47	120	21
14	9	39	9	71	4	96	63	121	18
15	15	40	5	72	6	97	37	122	12
16	13	41	8	73	1	98	34	123	6
17	4	42	6	74	6	99	45	124	6
18	14	43	12	75	4	100	36	125	3
19	6	44	11	76	5	101	25		
20	10	45	27	77	5	102	19		
21	10	46	28	78	4	103	36		
22	12	47	30	79	4	104	3		
23	11	48	26	80	7	105	33		
24	8	49	21	81	6	106	21		
25	16	50	18	82	7	107	22		
26	24	51	14	83	10	108	12		
27	28	52	7	84	2	109	21		
28	21	53	2	85	12	110	15		
29	28	54	3	86	11	111	9		
30	27	55	3	87	5	112	18		

STAGE 1 WING 1E2 TP-H1011 CONSTANT STRAIN

This sample size summary is applicable to figure 38

$Y = (( +2.5714980E+01) + (-1.7676117E-03) * X)$   
 $F = \text{SIGNIFICANCE OF } F = \text{NOT SIGNIFICANT}$   
 $R = \text{SIGNIFICANCE OF } R = \text{NOT SIGNIFICANT}$   
 $t = \text{SIGNIFICANCE OF } t = \text{NOT SIGNIFICANT}$   
 $N = 1597$   
 $\text{DEGREES OF FREEDOM} = 1597$   
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$



STAGE 1 WING 1&2 TP-H1011 CONSTANT STRAIN

Figure 38

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
88	3	117	15	142	24	196	6
91	3	118	24	143	33	197	2
92	3	119	42	144	18	198	3
93	9	120	29	145	15	202	3
94	9	121	20	146	3	202	2
95	9	122	15	147	6	204	3
97	7	123	13	148	3	206	6
98	7	124	18	150	3	206	12
99	9	125	5	152	16	207	9
101	9	126	21	153	12		
102	6	127	18	154	13		
103	21	128	24	155	19		
104	9	129	30	156	9		
105	5	130	30	157	3		
106	9	131	24	158	3		
107	24	132	9	159	3		
108	12	133	12	161	6		
109	24	134	24	164	6		
110	3	135	12	165	3		
111	5	136	24	167	6		
112	3	137	21	169	6		
113	24	138	30	190	6		
114	9	139	41	192	12		
115	21	140	9	193	9		
116	16	141	24	194	18		

WING 1E2 STAGE 1 TF-H1011 THERMAL COEFFICIENT OF LINEAR EXPANSION BELOW TG

This sample size summary is applicable to figures 39 and 40

$Y = (1 + 4.8504530E-05) + (6.0383324E-08) \times X$   
 $F = \text{SIGNIFICANCE OF } F = \text{SIGNIFICANT}$   
 $R = \text{SIGNIFICANCE OF } R = \text{SIGNIFICANT}$   
 $I = \text{SIGNIFICANCE OF } I = \text{SIGNIFICANT}$   
 $N = \text{DEGREES OF FREEDOM} = 1099$   
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = 5 DEGREES C/MIN

PARAMETER = TCLE BELOW T(G)  
 UNIT OF MEASURE = IN/IN/DEG C  $\times 10^{-3}$   
 0.02 0.04 0.06 0.08 0.10 0.12

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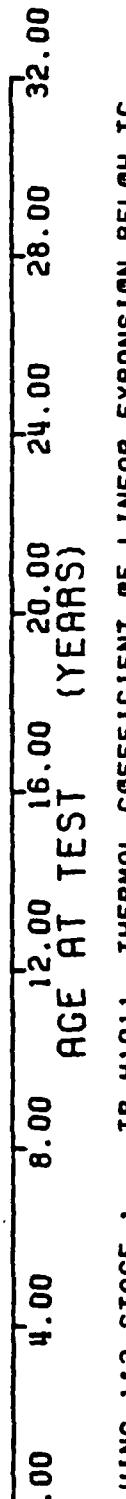
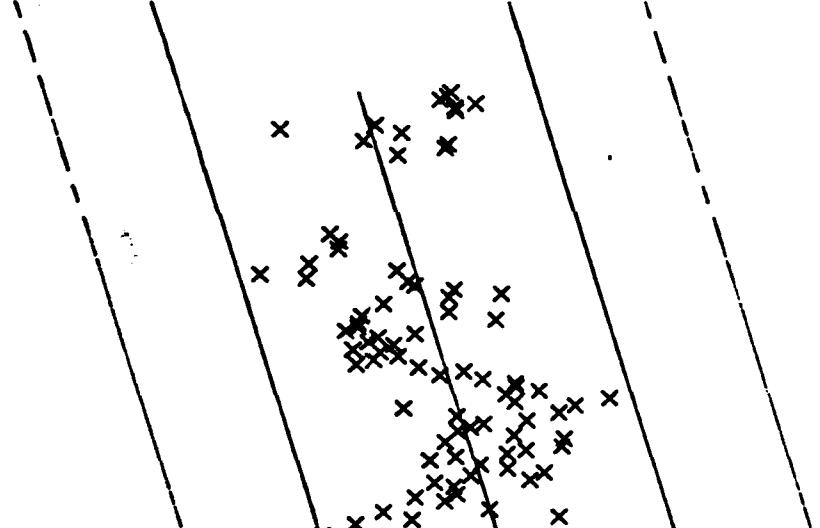


Figure 39

$\gamma = (1 + 7.0742642E-05) + (1 + 3203243E-07) * X$   
 $F = \text{SIGNIFICANCE OF } \gamma$   
 $R = \text{SIGNIFICANCE OF } \alpha$   
 $\alpha = \text{SIGNIFICANCE OF } \beta$   
 $N = \text{DEGREES OF FREEDOM} = 1099$   
 $N = \text{STORAGE CONDITIONS = AMB TEMP/RH}$

$\text{PARAMETER} = \text{TCLE ABOVE } T (G)$   
 $\text{UNIT OF MEASURE} = \text{IN/IN/DEG C} \times 10^{-3}$



$\text{WING 142 STAGE 1 TP-H1011. THERMAL COEFFICIENT OF LINEAR EXPANSION ABOVE } 16$   
 $\text{AGE AT TEST (YEARS)}$

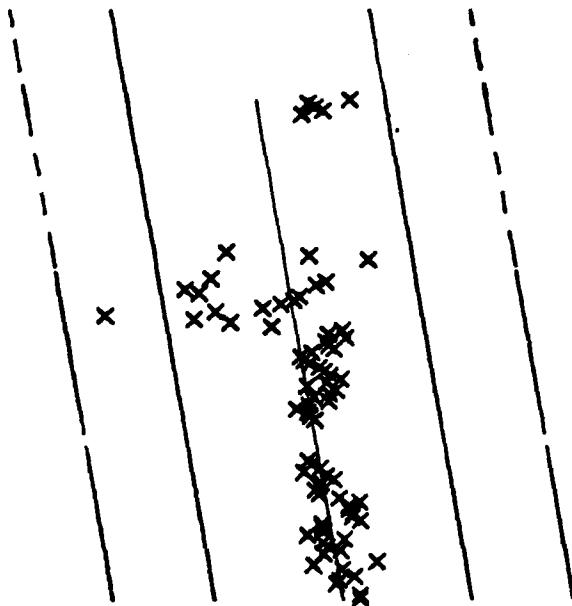
Figure 40

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
62	4	91	72	126	28	155	26
63	8	92	103	127	45	192	4
66	4	93	108	128	24	193	18
67	8	94	130	129	21	194	16
68	12	95	84	130	8	195	24
70	4	96	59	131	20	196	4
71	4	97	19	132	16		
72	3	98	24	133	12		
74	36	99	12	134	4		
75	6	110	28	135	8		
76	16	111	40	136	36		
77	20	112	32	137	32		
78	36	113	15	138	20		
79	20	114	36	139	28		
80	36	115	16	140	40		
81	56	116	24	141	28		
82	40	117	8	142	52		
83	24	118	16	143	40		
84	24	119	16	144	12		
85	24	120	28	145	12		
86	32	121	20	146	12		
87	20	122	56	147	4		
88	4	123	40	148	4		
89	72	124	28	153	12		
90	68	125	48	154	36		

$\gamma = ( +6.916648E-03 ) + ( +2.8884483E-05 ) \times X$   
 $F = 1873475E+02$  SIGNIFICANCE OF  $F =$  SIGNIFICANT  
 $R = +2.2197512E-01$  SIGNIFICANCE OF  $R =$  SIGNIFICANT  
 $R^2 = +1.0896547E+01$  SIGNIFICANCE OF  $R^2 =$  SIGNIFICANT  
 $N = 2293$  DEGREES OF FREEDOM = 2291  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

UNIT OF MEASURE = MILLIEV/C<sub>C</sub>  $\times 10^{-3}$   
 PARAMETER = CROSSLINK DENSITY



STAGE 1, WING 142, TP-H1011, SOL GEL, CROSSLINK DENSITY

Figure 41

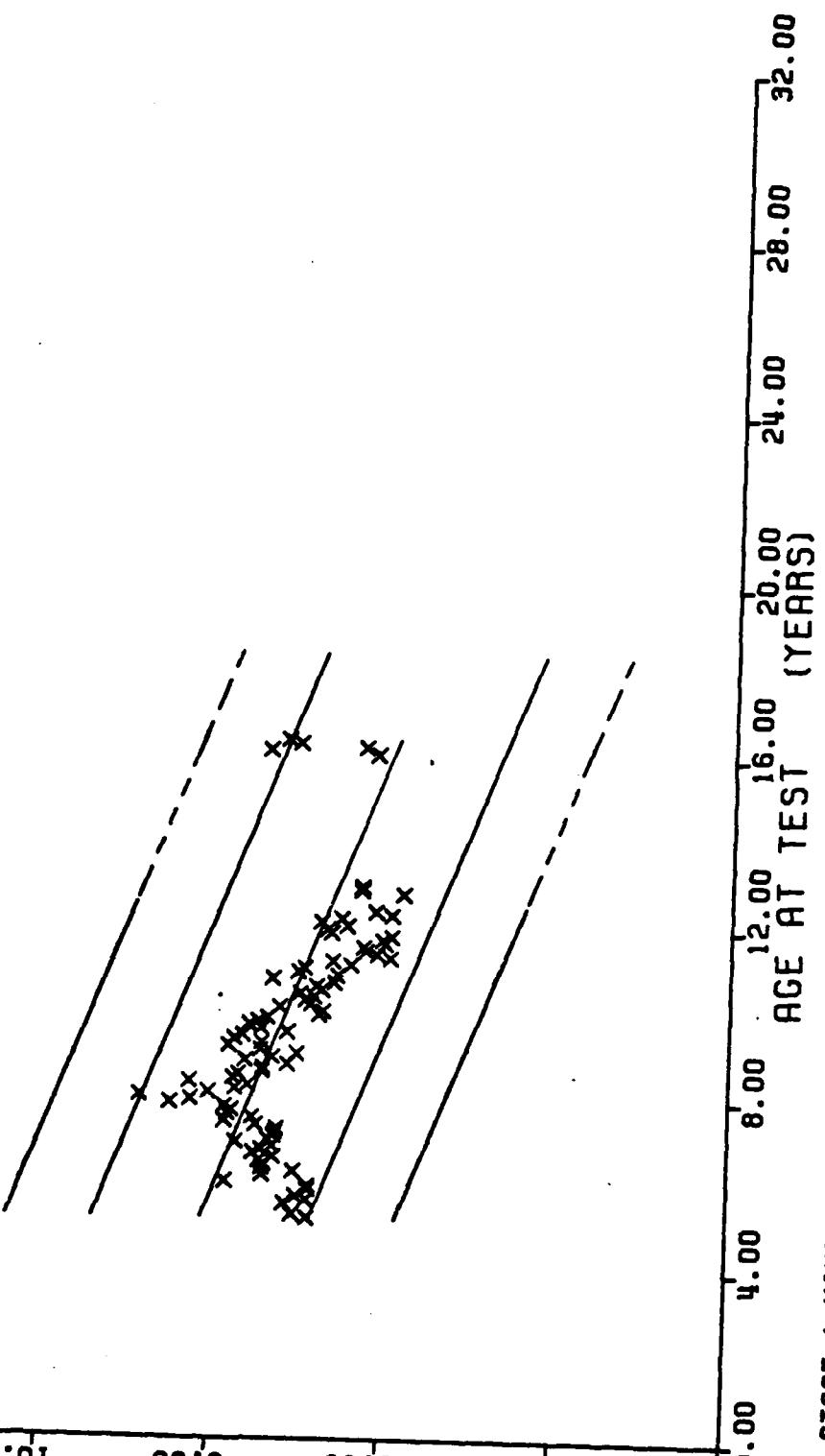
\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
62	4	91	72	116	40	141	28
63	8	92	103	117	12	142	52
66	4	93	108	118	32	143	39
67	8	94	130	119	16	144	12
68	12	95	84	120	28	145	12
70	4	96	59	121	20	146	12
71	4	97	15	122	56	147	4
72	3	98	24	123	40	148	4
74	36	99	12	124	28	153	12
75	8	100	16	125	48	154	36
76	16	101	32	126	27	155	26
77	20	102	16	127	44	192	4
78	36	103	8	128	24	193	18
79	20	104	12	129	21	194	16
80	36	105	12	130	8	195	24
81	56	106	20	131	20	196	4
82	40	107	32	132	16		
83	24	108	24	133	12		
84	24	109	32	134	4		
85	24	110	16	135	8		
86	32	111	24	136	36		
87	20	112	20	137	32		
88	4	113	20	138	19		
89	72	114	20	139	28		
90	68	115	16	140	40		

$\gamma = (+9.1480143E+00) + (-1.6515906E-02) \times x$   
 $\sigma_f = +8.7022567E-01$   
 $s_f = +5.8262065E-04$   
 $s_e = +7.5628466E-01$   
 $F = +8.0358733E+02$  SIGNIFICANCE OF  $F$  = SIGNIFICANT  
 $R = -4.9500198E-01$  SIGNIFICANCE OF  $R$  = SIGNIFICANT  
 $I = +2.8347616E+01$  SIGNIFICANCE OF  $I$  = SIGNIFICANT  
 $N = 2478$  DEGREES OF FREEDOM = 2476

STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

PARAMETER = % EXTRACTABLE  
 UNIT OF MEASURE = %



STAGE 1. WING 142. SOL GEL. % EXTRACTABLE, TPH-1011

Figure 42

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

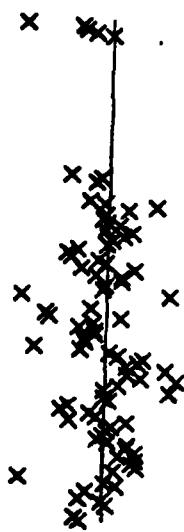
AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
62	4	91	72	116	48	141	28
63	8	92	103	117	12	142	52
66	4	93	108	118	32	143	44
67	8	94	130	119	16	144	12
68	12	95	84	120	28	145	12
70	4	96	59	121	20	146	12
71	4	97	19	122	56	147	4
72	3	98	24	123	40	148	4
74	36	99	12	124	28	153	12
75	8	100	16	125	48	154	36
76	16	101	32	126	27	155	28
77	20	102	16	127	44	192	4
79	36	103	8	128	24	193	16
79	20	104	12	129	21	194	16
80	36	105	12	130	8	195	24
81	56	106	20	131	20	196	4
82	40	107	32	132	16		
83	24	108	24	133	12		
84	24	109	36	134	4		
85	24	110	28	135	8		
86	32	111	40	136	36		
87	20	112	32	137	32		
88	4	113	35	138	19		
89	72	114	40	139	28		
90	68	115	28	140	40		

STAGE 1. WING 162. TP-H1011. SOL GEL. WT. SWELL RATIO

This sample size summary is applicable to figure 43

$Y = (( +4.0794532E+00) + (-3.8636605E-04) * X)$   
 $F = \text{SIGNIFICANCE OF } F = \text{NOT SIGNIFICANT}$   
 $R = \text{SIGNIFICANCE OF } R = \text{NOT SIGNIFICANT}$   
 $I = \text{SIGNIFICANCE OF } I = \text{NOT SIGNIFICANT}$   
 $N = 2578 \text{ DEGREES OF FREEDOM} = 2576$   
 TEST CONDITIONS = AMB TEMP/RH  
 STORAGE CONDITIONS = AMB TEMP/RH

PARAMETER = WEIGHT SWELL  
 UNIT OF MEASURE = RATIO  
 6.40 6.60 6.80 7.00 7.20 7.40  
 2.40 2.60 2.80 3.00 3.20 3.40



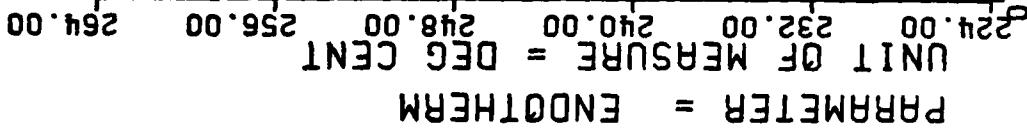
STAGE 1. WING 1&2. TP-H1011. SOL GEL. WT. SWELL RATIO

Figure 43

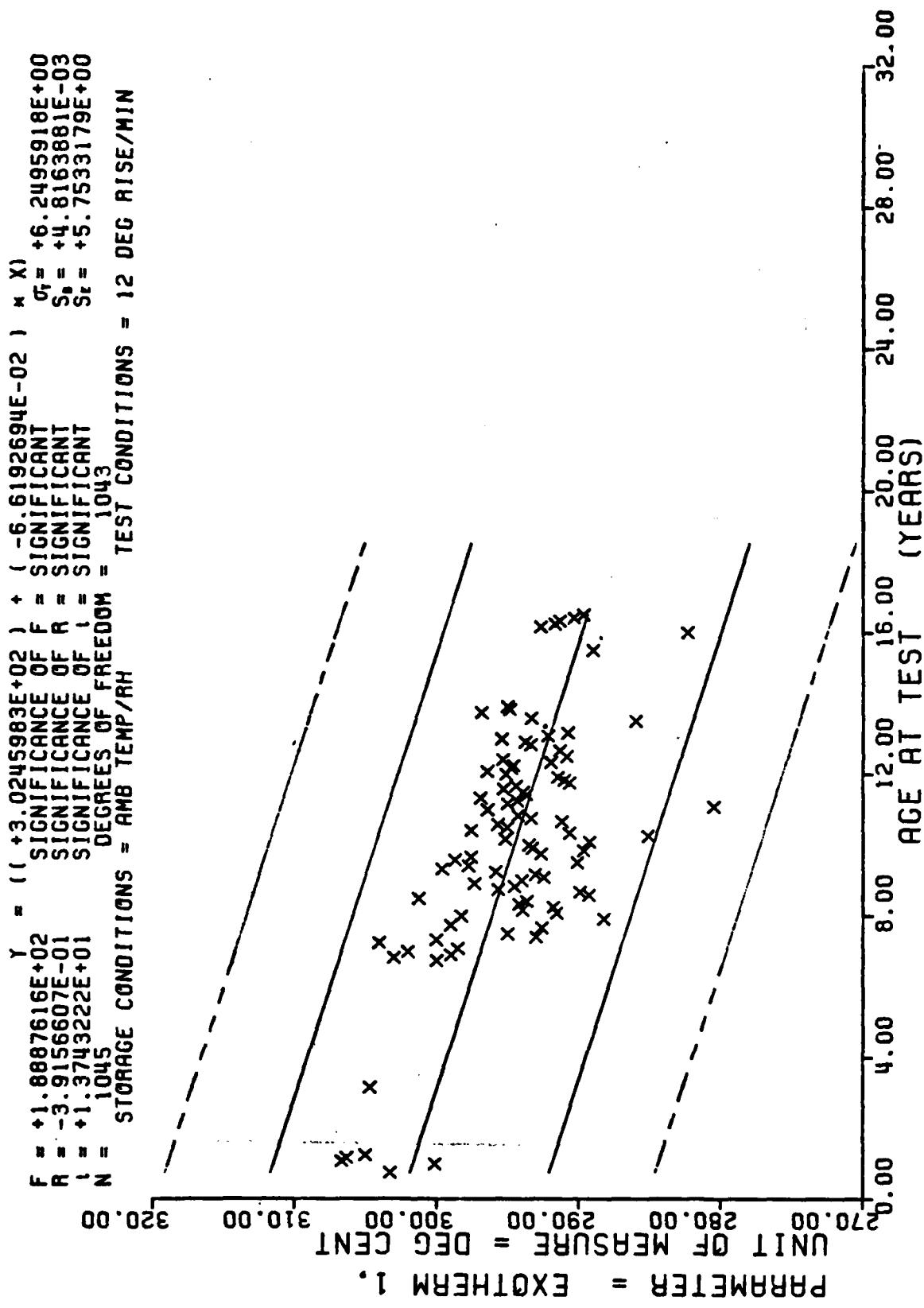
\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MCS)	NR SAMPLE	AGE (MCS)	NR SAMPLE	AGE (MCS)	NR SAMPLE	AGE (MCS)	NR SAMPLE
9	16	103	21	128	6	156	12
12	24	104	12	129	3	157	6
13	12	105	6	130	4	158	5
14	16	106	27	132	5	162	3
15	8	107	5	133	4	163	2
16	3	108	9	134	16	165	6
17	1	109	11	135	25	166	6
18	1	110	23	126	25	167	5
19	1	111	11	137	29	168	5
20	1	112	17	138	43	192	5
21	2	113	34	139	38	194	5
22	1	114	11	140	21	195	6
23	1	115	19	141	8	196	6
24	1	116	18	142	8	197	6
25	5	117	14	143	22	198	6
26	5	118	17	144	11		
27	5	119	24	145	12		
28	4	120	18	146	21		
29	4	121	21	147	12		
30	12	122	6	148	17		
31	16	123	9	149	3		
32	21	124	17	150	6		
33	9	125	11	152	4		
34	17	126	7	154	24		
35	14	127	12	155	21		
36	9						

$\gamma = (1 + 2.492488E+02) + (-4.9712235E-02) \times X$   
 $F = \text{SIGNIFICANT}$   
 $\sigma_f = +3.1032203E+00$   
 $\alpha = -5.9114976E-01$   
 $F = \text{SIGNIFICANT}$   
 $S_0 = +2.0961873E-03$   
 $\alpha = +2.3715550E+01$   
 $F = \text{SIGNIFICANT}$   
 $S_t = +2.5041340E+00$   
 $N = 1049$   
 $\text{DEGREES OF FREEDOM} = 1047$   
 $\text{TEST CONDITIONS} = 12 \text{ DEG RISE/MIN}$   
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$



STAGE 1. WING 142. DTA. 12 DEG C RISE/MIN.. ENDOOTHERM



STAGE 1. WING 142.01A.12 DEG C RISE/MIN. EXOTHERM #1

Figure 45

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
9	16	103	18	128	6	156	9
12	24	104	10	129	3	157	6
13	12	105	4	130	4	158	9
14	16	106	23	132	5	162	7
15	8	107	8	133	4	163	1
16	3	108	5	134	16	165	6
18	1	109	8	135	23	166	6
81	1	110	18	136	25	167	6
82	1	111	11	137	29	186	1
83	1	112	15	138	42	192	1
84	1	113	34	139	37	194	6
85	2	114	10	140	20	195	6
87	1	115	19	141	8	196	1
88	1	116	18	142	8	197	6
89	1	117	12	143	20	198	1
90	4	118	13	144	11		
92	5	119	22	145	12		
93	4	120	18	146	21		
95	4	121	20	147	12		
96	6	122	6	148	17		
97	10	123	8	149	3		
99	14	124	16	150	6		
100	12	125	11	152	6		
101	13	126	3	154	20		
102	9	127	12	155	17		

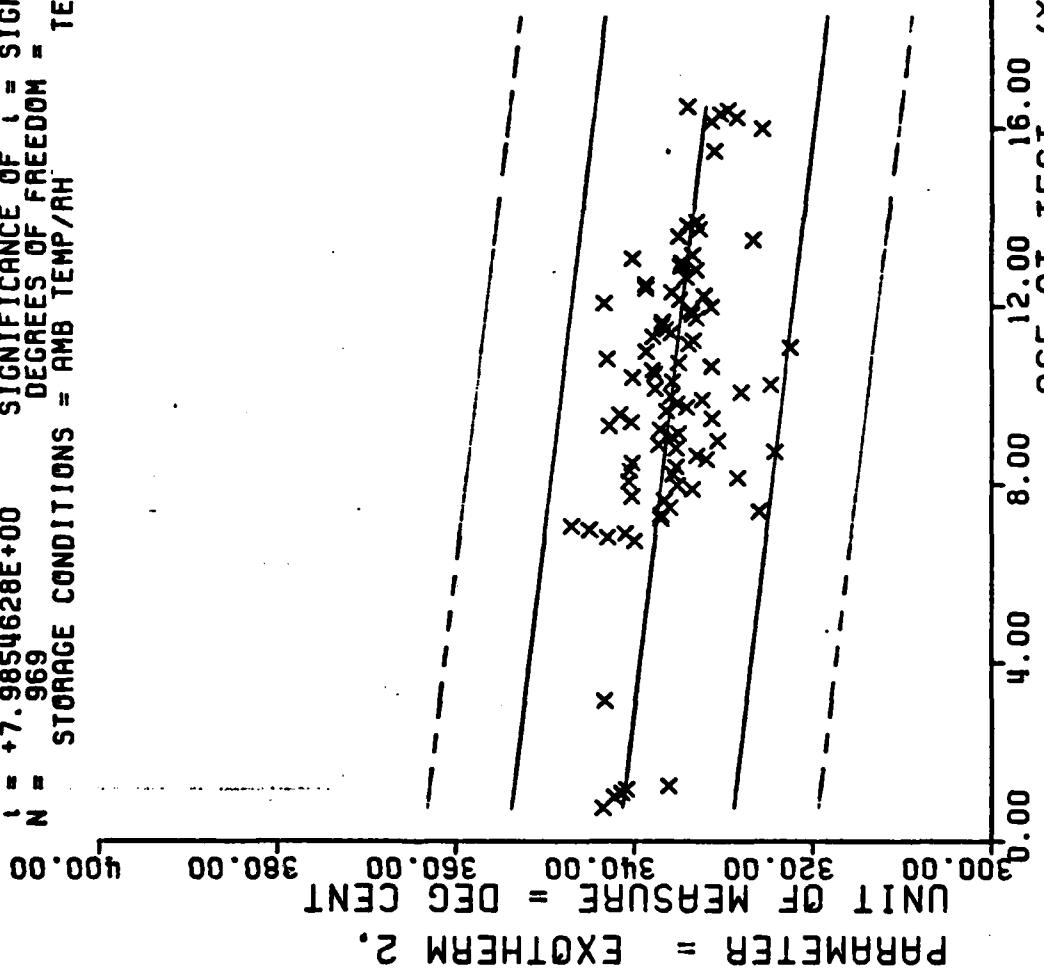
STAGE 1. WING 162.DTA.12 DEG C RISE/MIN. EXOTHERM #2

This sample size summary is applicable to figure 46

$F = +6.3767616E+01$       SIGNIFICANCE OF  $F =$  SIGNIFICANT  
 $R = -2.4872515E-01$       SIGNIFICANCE OF  $R =$  SIGNIFICANT  
 $I = +7.9854628E+00$       SIGNIFICANCE OF  $I =$  SIGNIFICANT  
 $N = 969$       DEGREES OF FREEDOM = 967

STORAGE CONDITIONS = AMB TEMP/RH

TEST CONDITIONS = 12 DEG RISE/MIN



\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
85	1	120	7	163	1
90	1	121	2		
92	2	123	3		
95	3	124	10		
96	2	125	1		
97	3	127	5		
98	3	128	6		
99	5	135	11		
100	2	136	9		
101	7	137	12		
103	7	138	19		
104	5	139	19		
106	5	140	10		
107	4	141	6		
109	7	142	8		
110	13	143	9		
111	5	144	8		
112	7	145	8		
113	11	146	6		
114	4	147	2		
115	5	148	8		
116	2	150	5		
117	5	152	2		
118	11	154	1		
119	10	158	5		

STAGE 1, WING 1&2, DTA, 12 DEG C RISE/MIN. EXOTHERM #3

This sample size summary is applicable to figure 47

$F = +9.4272102E+00$   
 $R = +1.7152479E-01$   
 $t = +3.0703762E+00$   
 $N = 313$   
 $F = (( +3.5734796E+02 ) + ( +8.4408252E-02 ) * X) / ( +8.2720124E+00 )$   
 $R = +2.7491175E-02$   
 $t = +8.1625110E+00$   
 $DEGREES OF FREEDOM = 311$   
 $STORAGE CONDITIONS = AMB TEMP/RH$   
 $TEST CONDITIONS = 12 DEG RISE/MIN$

$UNIT OF MEASURE = DEG CENT$   
 $PARAMETER = EXOTHERM 3,$

STAGE 1, WING 1&2, DTA, 12 DEG C RISE/MIN, EXOTHERM #3

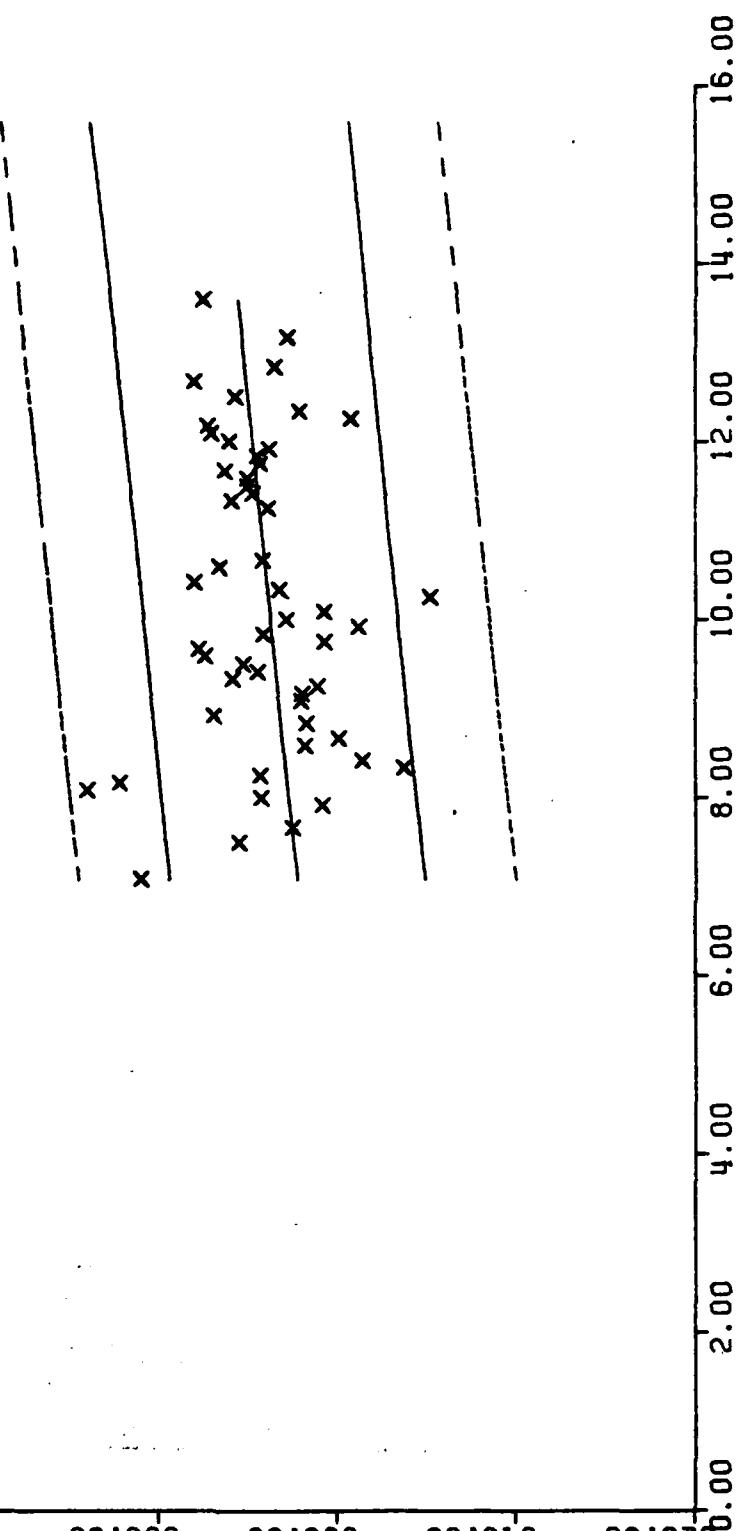


Figure 47

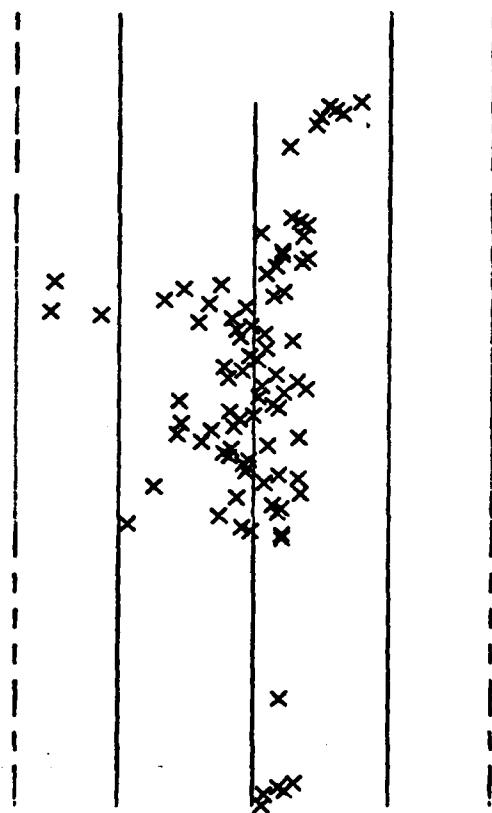
*** SAMPLE SIZE SUMMARY ***					
AGE (WOS)	NR SAMP	AGE (WNS)	NR SAMP	AGE (WOS)	NR SAMP
9	16	103	20	129	3
12	24	104	11	130	2
13	12	105	6	132	5
14	16	106	23	134	14
15	8	107	9	135	16
38	3	108	9	136	15
81	1	109	7	137	19
82	1	110	15	138	27
83	1	111	7	139	25
84	1	112	14	140	11
85	2	113	23	141	6
87	1	114	8	142	2
88	1	115	17	143	15
89	1	116	16	144	6
90	5	117	10	145	6
92	4	118	11	146	14
93	4	119	16	147	9
95	2	120	11	148	17
96	12	121	20	149	3
97	13	122	2	150	4
98	20	123	7	152	4
99	6	124	9	154	24
100	15	125	10	155	21
101	8	126	3	156	12
102	5	127	8	157	6

STAGE 1, LING 162, STA, 12 DEG C RISE/MIN, IGNITION

This sample size summary is applicable to figure 48

$F = +1.0078698E-01$   
 $R = -1.1112975E-02$   
 $t = +3.1746966E-01$   
 $N = 818$   
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$   
 $\text{DEGREES OF FREEDOM} = 816$   
 $\text{TEST CONDITIONS} = 12 \text{ DEG RISE/MIN}$

$\text{PARAMETER} = \text{IGNITION TEMP}$   
 $\text{UNIT OF MEASURE} = \text{DEG CENT}$   
 $0.00 \quad 400.00 \quad 320.00 \quad 360.00 \quad 400.00 \quad 440.00 \quad 480.00$



STAGE 1. WING 1&2, DTA, 12 DEG C RISE/MIN, IGNITION

Figure 48

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
84	2	112	9	137	47	169	3
87	1	113	11	138	24	170	6
88	2	114	4	139	17	171	5
89	3	115	25	140	26	172	3
91	3	116	15	141	12	174	3
92	1	117	17	142	24	179	9
93	3	118	23	143	16	183	5
94	2	119	18	144	7	189	5
95	2	120	19	145	3	191	12
96	1	121	12	146	4	192	3
97	5	122	10	147	2	193	15
98	5	123	36	148	2	194	3
99	8	124	15	149	3	195	6
100	7	125	30	153	3	197	6
101	10	126	24	154	6	199	5
102	8	127	35	155	15	200	5
103	5	128	30	156	24		
104	10	129	27	157	3		
105	6	130	21	158	3		
106	7	131	27	160	3		
107	11	132	15	161	3		
108	16	133	11	163	6		
109	16	134	14	165	6		
110	4	135	19	166	6		
111	4	136	29	167	6		

STAGE 1, WING 162, PRESSURE TIME, MAXIMUM PRESSURE

This sample size summary is applicable to figures 49 and 50

STORAGE CONDITIONS = AMB TEMP/RH      TEST CONDITIONS = TEST COND 500 PS

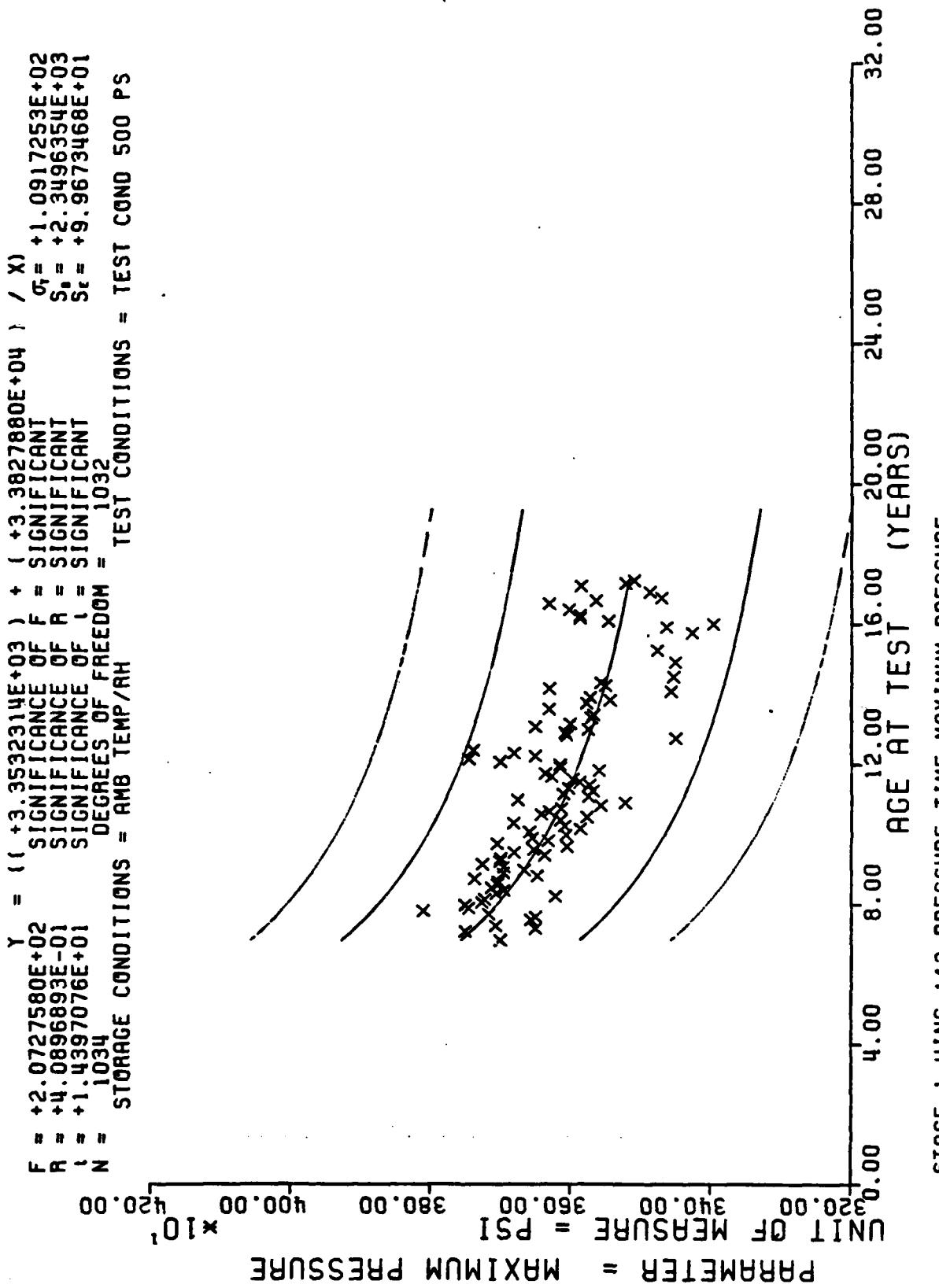


Figure 49

$F = +6.4639742E+01$        $Y = (( +5.9246098E-01 ) + ( +4.9137723E-04 ) * X)$   
 $R = +2.4289351E-01$       SIGNIFICANCE OF F = SIGNIFICANT  
 $t = +8.0398844E+00$       SIGNIFICANCE OF R = SIGNIFICANT  
 $N = 1033$       SIGNIFICANCE OF I = SIGNIFICANT  
DEGREES OF FREEDOM = 1031  
STORAGE CONDITIONS = AMB TEMP/RH      TEST CONDITIONS = TEST COND 500 PS

PARAMETER = TIME TO MAX PRESS      UNIT OF MEASURE = SECONDS  
0.00      0.56      0.64      0.72      0.80      0.88

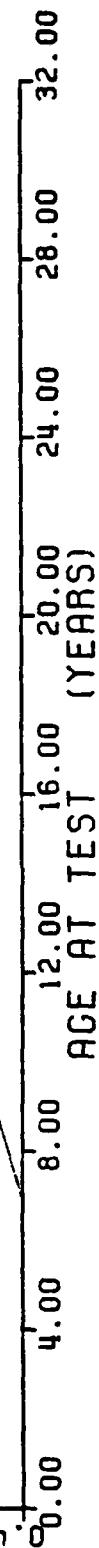


Figure 50

AD-A093 864

OGDEN AIR LOGISTICS CENTER HILL AFB UT PROPELLANT AN--ETC F/G 21/9.2  
PROPELLANT SURVEILLANCE REPORT LGM-30 A & B STAGE I, TP-H1011. (U)  
NOV 80 J A THOMPSON  
MAKPH-448(80)

UNCLASSIFIED

NI

1 of 2  
87-864-2

END  
0000  
FILED  
2-3A  
DTIC

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

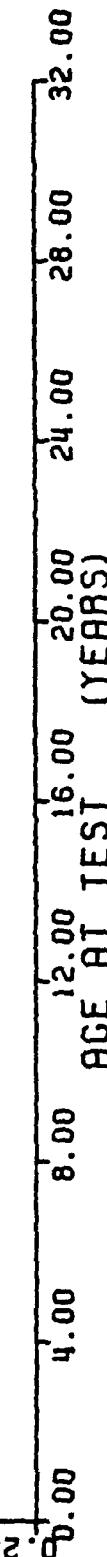
AGE (MCS)	NR SAMP	AGE (MCS)	NR SAMP	AGE (MCS)	NR SAMP	AGE (MCS)	NR SAMP
78	21	105	21	130	24	157	16
80	18	106	18	131	21	158	3
81	21	107	21	132	12	155	6
82	18	108	18	133	26	164	7
83	15	109	15	134	36	165	6
85	9	110	9	135	9	166	6
86	24	111	24	136	51	167	6
87	27	112	27	137	45	168	6
88	21	113	21	138	33	169	6
89	21	114	21	139	48	188	3
90	18	115	18	140	18	192	18
91	18	116	18	141	18	194	6
92	45	117	45	142	9	195	6
93	21	118	21	143	9	196	9
94	18	119	18	144	9	197	9
95	24	120	24	145	6	200	6
96	18	121	18	146	3	202	6
97	18	122	18	148	9	206	6
98	26	123	26	149	3	208	6
99	21	124	21	150	7	207	9
100	24	125	24	152	6	208	16
101	35	126	35	153	15		
102	22	127	22	154	6		
103	36	128	36	155	12		
104	29	129	29	156	15		

STAGE 1, WING ACB, 1F-H1011, BURNING RATE 1000 PSI

This sample size summary is applicable to figure 51

$\gamma = (( +3.4582381E-01) + (-2.8139280E-04) \times X)$   
 $F = +3.1570158E+02$  SIGNIFICANCE OF F = SIGNIFICANT  
 $R = -4.3263261E-01$  SIGNIFICANCE OF R = SIGNIFICANT  
 $t = +1.7767993E+01$  SIGNIFICANCE OF t = SIGNIFICANT  
 $N = 1373$  DEGREES OF FREEDOM = 1371  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = 1000 PSI

PARAMETER = BURNING RATE  
 UNIT OF MEASURE = IN/SEC  
 0.20 0.24 0.28 0.32 0.36 0.40



STAGE 1. WING A4B. TP-H1011. BURNING RATE 1000 PSI

Figure 51

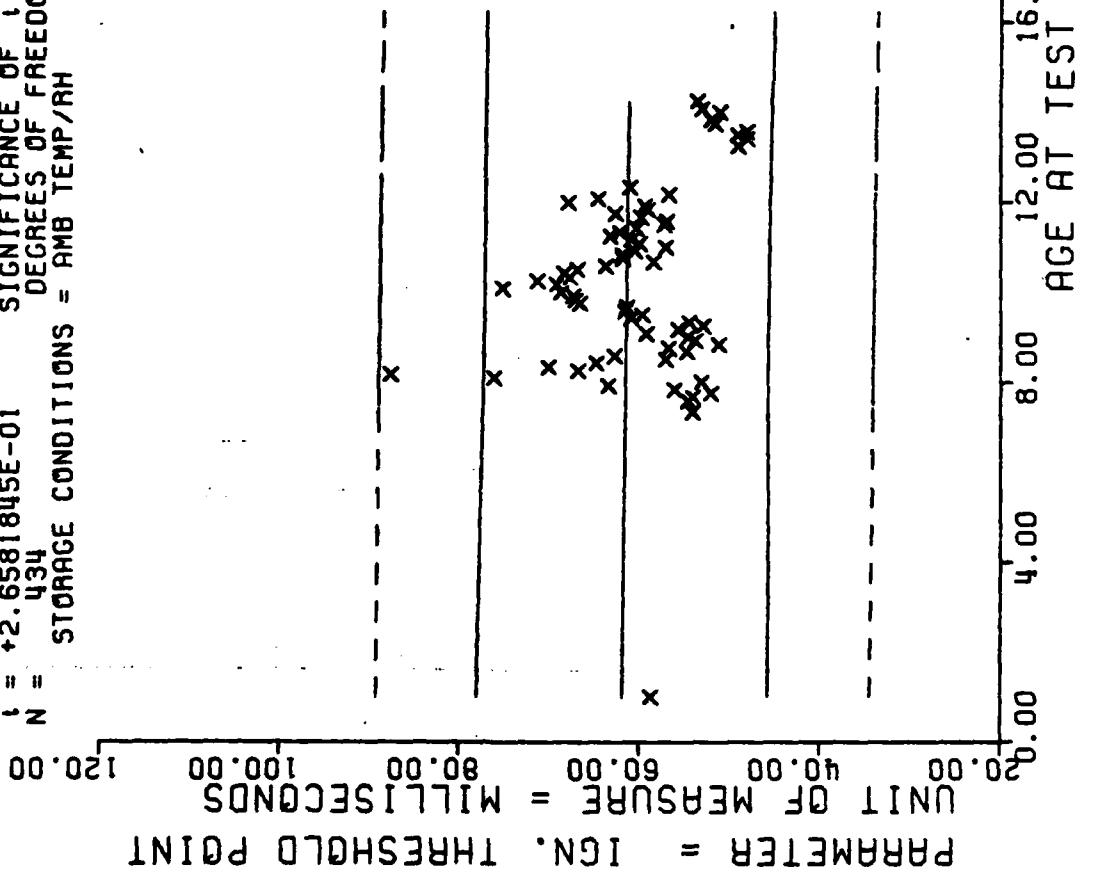
\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
88	1	114	12	139	11		
91	2	115	6	140	4		
92	4	116	10	141	5		
93	1	117	6	142	5		
94	3	118	6	143	7		
95	3	119	12	144	5		
96	3	120	8	145	2		
97	4	121	8	146	3		
98	4	122	9	148	2		
99	10	123	12	159	1		
100	7	124	5	161	1		
101	10	125	11	162	1		
102	8	126	9	163	1		
103	9	127	11	165	2		
104	9	128	8	166	2		
105	6	129	4	168	1		
106	10	130	6	169	1		
107	7	131	7	171	2		
108	12	132	5				
109	6	133	9				
110	8	134	11				
111	8	135	8				
112	12	136	8				
113	7	137	8				
		138	15				

STAGE 1 WING 162 TP-H1011 IGNITABILITY. IGN THRESHOLD POINT. 168 CAL/SD CM/SEC

This sample size summary is applicable to figure 52

$Y = (( +6.1921518E+01) + (-5.0982735E-03) * X)$   
 $F = +7.0659453E-02$  SIGNIFICANCE OF  $F = \text{NOT SIGNIFICANT}$   $\sigma_f = +9.1103220E+00$   
 $R = -1.2788150E-02$  SIGNIFICANCE OF  $R = \text{NOT SIGNIFICANT}$   $S_f = +1.9179531E-02$   
 $t = +2.6581845E-01$  SIGNIFICANCE OF  $t = \text{NOT SIGNIFICANT}$   $S_t = +9.1201144E+00$   
 $N = 434$  DEGREES OF FREEDOM = 432  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = 168 CAL/SQCM/SEC



\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
48	3	140	23	195	3
105	12	141	15	196	6
106	6	142	9	197	3
108	24	143	3	198	3
110	12	144	3	199	6
112	12	145	5	200	3
114	6	146	7	202	9
115	12	147	6		
116	15	148	3		
120	5	156	3		
121	9	159	9		
124	16	163	3		
125	15	164	4		
126	3	165	4		
127	21	166	4		
128	19	167	6		
129	11	168	5		
130	9	174	2		
131	9	176	7		
132	12	177	6		
135	7	188	3		
136	11	190	3		
137	14	192	10		
138	6	193	11		
139	22	194	15		

STAGE I WING 1C2 DIFFERENTIAL SCANNING CALORIMETER ENDOTHERM PEAK TEMP

This sample size summary is applicable to figures 53, 54 and 55

$F = +2.9972406E+00$        $Y = (( +5.1809439E+02 ) + (-4.4738058E-03 ) \times X)$   
 $F =$  SIGNIFICANCE OF  $F$  = NOT SIGNIFICANT       $\sigma_t = +1.6468736E+00$   
 $R = -7.8210306E-02$        $F =$  SIGNIFICANCE OF  $R$  = NOT SIGNIFICANT       $S_t = +2.5841417E-03$   
 $R = +1.7312540E+00$        $R =$  SIGNIFICANCE OF  $R$  = NOT SIGNIFICANT       $S_r = +1.6435138E+00$   
 $N = 489$       DEGREES OF FREEDOM = 487  
 $N =$  STORAGE CONDITIONS = AMB TEMP/RH      TEST CONDITIONS = AMB TEMP/RH

UNIT OF MEASURE = DEG K  
 PARAMETER = ENDOOTHERM PEAK TEMP  
 508.00 512.00 516.00 520.00 524.00 528.00

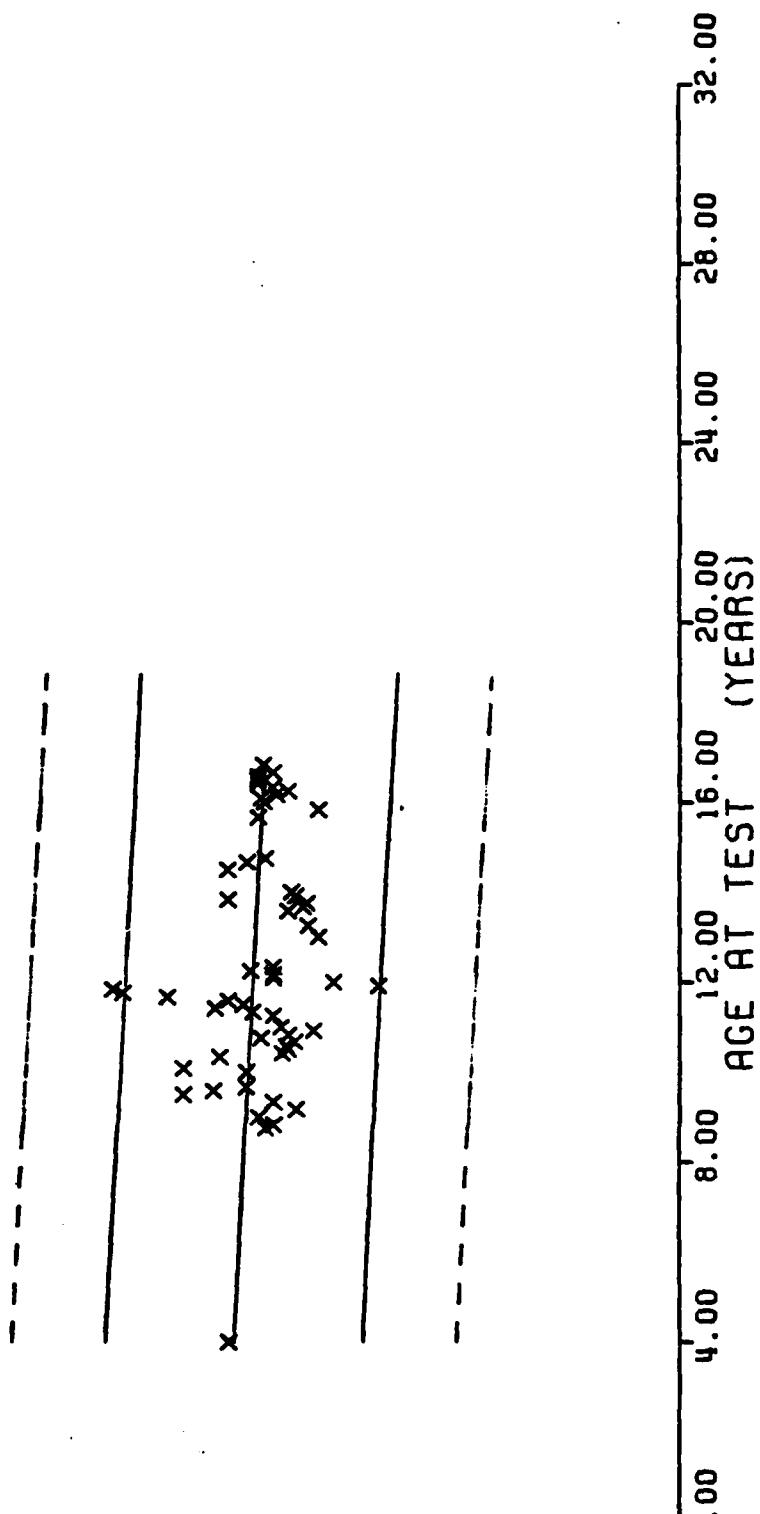


Figure 53

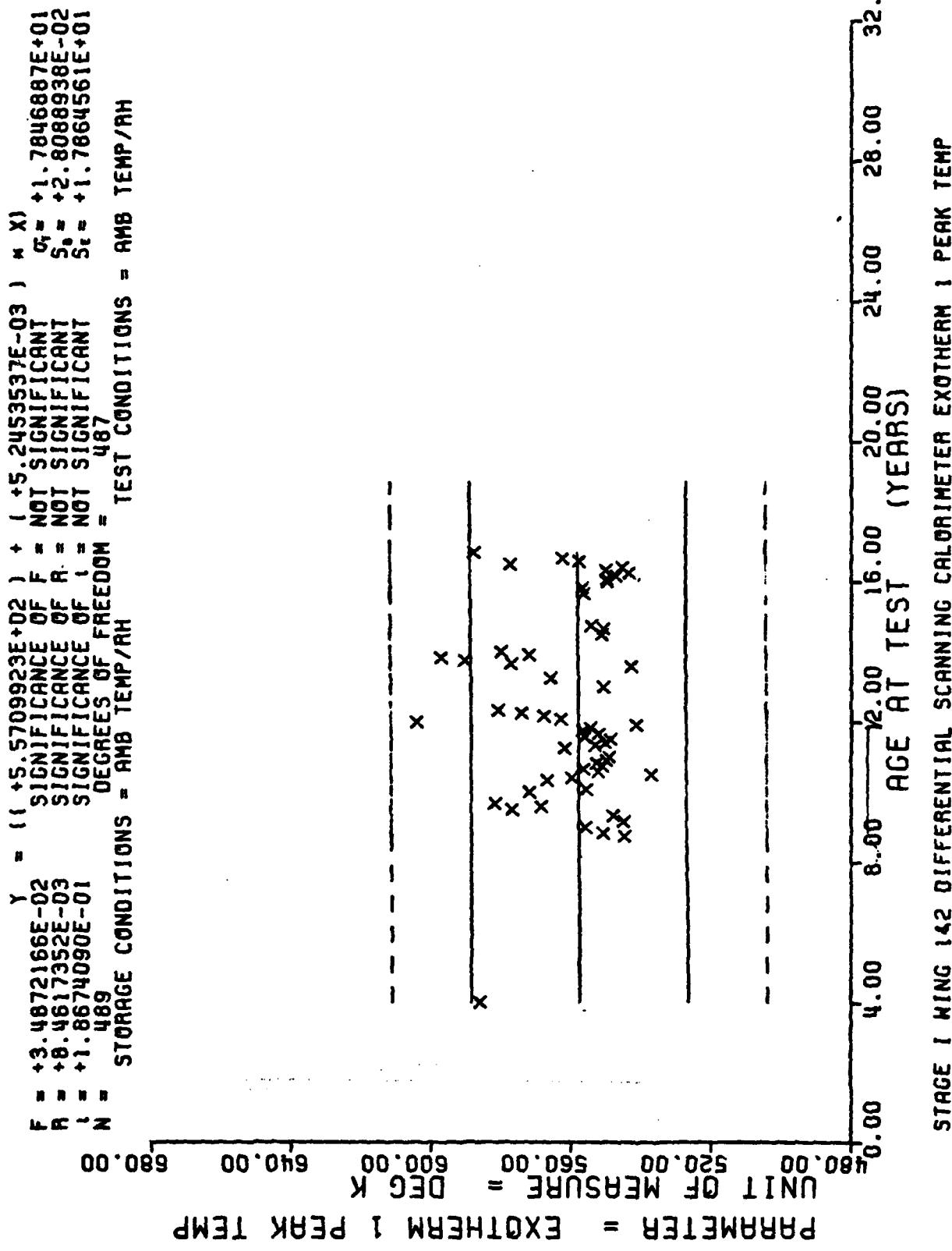
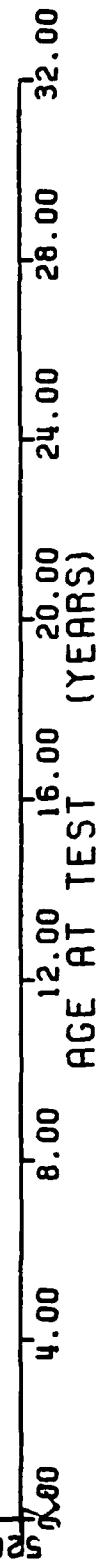


Figure 54

$F = 2.6216645E+00$   
 $F = 7.3249083E-02$   
 $F = 1.6191554E+00$   
 $F = 4.88$   
 $N = 142$   
 $\text{Y} = ( ( +5.9905637E+02 ) + ( +5.1452111E-02 ) * X )$   
 $\text{Y} = ( ( +5.9905637E+02 ) + ( +5.1452111E-02 ) * X )$   
 $F = \text{NOT SIGNIFICANT}$   
 $F = \text{NOT SIGNIFICANT}$   
 $F = \text{NOT SIGNIFICANT}$   
 $F = \text{NOT SIGNIFICANT}$   
 $Degrees of Freedom = 486$   
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$

$\text{PARAMETER} = \text{EXOTHERM 2 PER K TEMP}$   
 $\text{UNIT OF MEASURE} = \text{DEG K}$   
 $520.00 560.00 600.00 640.00 680.00 720.00$



STAGE I WING 142 DIFFERENTIAL SCANNING CALORIMETER EXOTHERM 2 PEAK TEMP

Figure 55

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (WCS)	NR SAMP	AGE (MDS)	NR SAMP	AGE (MDS)	NR SAMP
9	6	127	1	168	6
12	4	128	6	203	4
82	1	129	5	207	2
84	1	130	6	208	12
85	1	131	6	209	12
89	2	132	5	4	2
91	4	133	4	2	2
92	1	134	10	2	2
97	6	135	14	2	2
58	5	136	14	2	2
99	4	137	8	2	2
100	5	138	4	2	2
101	1	139	4	2	2
102	3	153	4	2	2
103	3	154	4	2	2
105	1	155	4	2	2
106	1	156	4	2	2
107	1	186	4	2	2
112	2	190	7	2	2
113	2	191	4	2	2
114	6	192	4	2	2
115	2	193	11	2	2
116	3	194	4	2	2
125	2	195	8	2	2
126	1	196	2	2	2

WING 162 STAGE 1 TGA FERCENT WEIGHT LOSS AT IGNITION. 9 DEG C RISE/MIN

This sample size summary is applicable to figures 56 and 57

$F = +6.4932960E+01$   
 $F = +2.5913622E+01$   
 $F = +7.7079085E-02$   
 SIGNIFICANCE OF F = SIGNIFICANT  
 $R = +4.4771849E-01$   
 $R = +8.0580990E+00$   
 $R = +7.0854209E+00$   
 $R = +9.5654180E-03$   
 $R = +7.0854209E+00$   
 SIGNIFICANCE OF R = SIGNIFICANT  
 $I = +8.261$   
 DEGREES OF FREEDOM = 259  
 TEST CONDITIONS = 9 DEG C RISE/MIN  
 STORAGE CONDITIONS = AMB TEMP/RH  
 $N = 261$

UNIT OF MEASURE = PERCENT  
 PARAMETER = WEIGHT LOSS  
 100.00  
 80.00  
 60.00  
 40.00  
 20.00  
 0.00

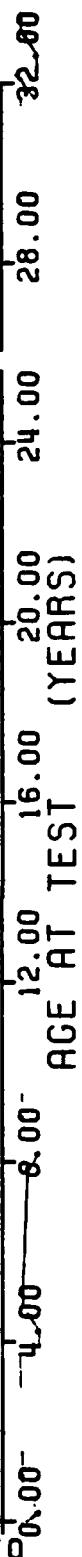


Figure 56

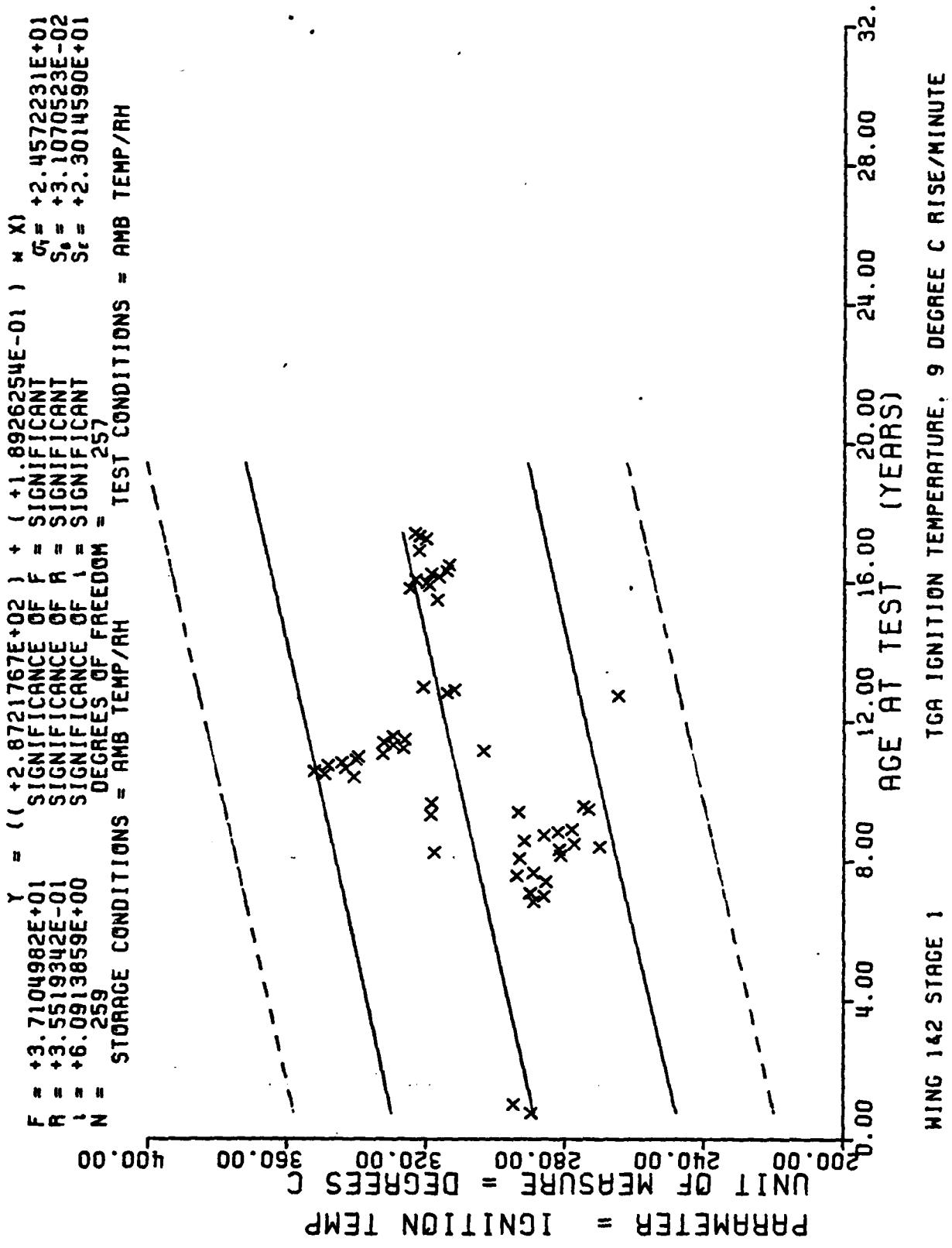


Figure 57

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

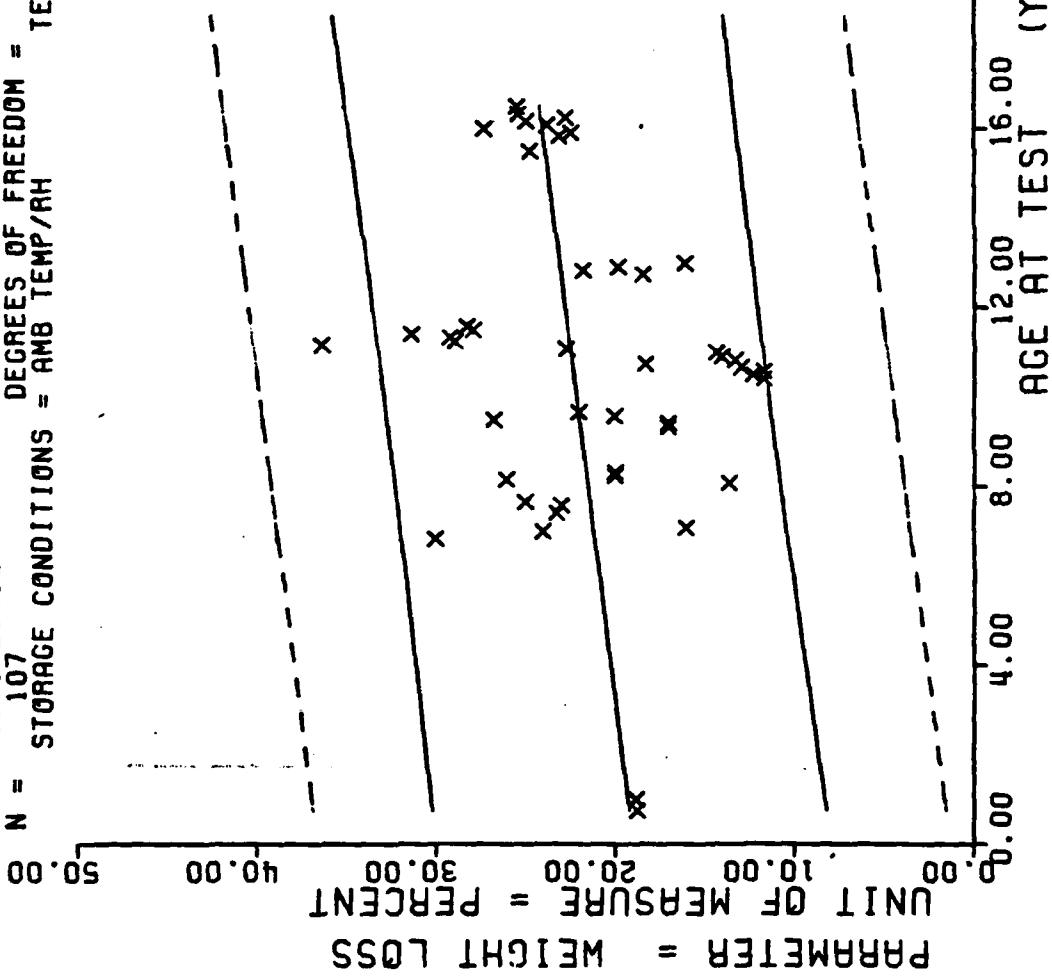
AGE (NCS)	NR SAMPLE	AGE (NCS)	NR SAMPLE
9	6	133	2
12	4	134	5
82	1	135	4
84	1	136	6
85	1	137	1
89	2	138	2
51	2	139	2
92	1	163	4
97	4	164	2
58	7	155	2
99	2	156	2
100	1	166	2
112	1	190	1
113	1	191	4
114	2	192	2
115	1	193	6
116	1	194	2
125	1	195	4
126	2	196	1
127	1	198	2
128	2	203	2
129	1	207	1
120	1	208	6
131	2	209	6
132	2		

WING 132 STAGE 1

X WT LCSS A1 250 DEG C HCLC, 12 DEG RISE/MIN TO HOLD

This sample size summary is applicable to figure 58

$F = +5.7274501E+00$        $y = (( +1.8970762E+01 ) + ( +2.6565350E-02 ) * x)$   
 $R = +2.2743275E-01$        $F = \text{SIGNIFICANT}$   
 $R = +2.3932091E+00$        $F = \text{SIGNIFICANT}$   
 $t = +2.3932091E+00$        $F = \text{SIGNIFICANT}$   
 $N = 107$        $\text{DEGREES OF FREEDOM} = 105$   
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$        $\text{TEST CONDITIONS} = 12 \text{ DEG R/M TO HOLD}$



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20. ABSTRACT (Continue on reverse side if necessary and identify by block number)  This report contains propellant test results from cartons of TP-H1011 bulk propellant representing LGM-30 A & B First Stage Minuteman Motors. Testing was accomplished in accordance with MMWRBM Project M04046C-WNL01529. The purpose of testing was to determine and provide early warning of any serious degradation trends occurring in the propellant for service life predictions.		

An analysis of all parameters indicates that no potential problems are expected in the propellant for at least two years past the oldest data point.

Data stored in the GO85 System were plotted utilizing the IBM 360-65 Computer and CAL-COMP Plotter. The data range at any age can be found by suitable inquiry of the GO85 System.

Each point on the regression plot represents the mean of all samples at that particular age. The number of specimens at each point is indicated on the sample size summary sheet accompanying each regression plot or group of regression plots.